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# LM RENDEZVOUS PROCEDURES

## G MISSION

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LM

RENDEZVOUS PROCEDURES

G MISSION

AS-506/CSM-107/LM-5

May 16, 1969

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LIST OF ACRONYMS AND ABBREVIATIONS

ACQ	Acquisition
ADJ	Adjust
AGS	Abort Guidance System
AOH	Apollo Operations Handbook
AOS	Acquisition of Signal
AOT	Alignment Optical Telescope
APS	Ascent Propulsion System
ATT	Attitude
BU	Backup
CALIB	Calibration
CB	Circuit Breaker
CDH	Constant Delta Height
CDR	Commander
CM	Command Module
CMC	Command Module Computer
CMD	Command Module Commander's Position
CMP	Command Module Pilot
COAS	Crew Optical Alignment Sight
CSI	Concentric Sequence Initiation
CSM	Command and Service Module
CT	Cease Tracking
DAP	Digital Autopilot
DH	Delta Height
DOI	Descent Orbit Insertion
DPS	Descent Propulsion System
DSKY	Display and Keyboard
DV	Delta Velocity
DWN	Down
EMS	Entry Monitor System
ET	Event Timer
FDAI	Flight Director Attitude Indicator
FPS	Feet Per Second
FWD	Forward
GDC	Gyro Display Coupler
GET	Ground Elapsed Time
GETI	Ground Elapsed Time of Ignition
GMBL	Gimbal
GND	Ground (Mission Control)
HA	Apogee Altitude
HGA	High-Gain Antenna
HOR	Horizon
HORIZ	Horizontal
HP	Perigee Altitude
IMU	Inertial Measurement Unit
INS	Insertion
IT	Initiate Tracking
LGC	LM Guidance Computer
LM	Lunar Module

LMP	Lunar Module Pilot
LOI	Lunar Orbit Injection
LOS	Line Of Sight
LV	Launch Vehicle
MNVR	Maneuver
MCC	Midcourse Correction
MCC1	First Midcourse Correction
MCC2	Second Midcourse Correction
MCC-H	Mission Control Center - Houston
MGA	Middle Gimbal Angle
MSFN	Manned Space Flight Network
NAV	Navigation
ORDEAL	Orbital Rate Drive Earth and Lunar
PAD	Data Voiced to Crew From Ground
PB	Pushbutton
PC	Plane Change
PDI	Powered Descent Initiation
PGNS	Primary Guidance, Navigation, and Control System
PHS	Phasing
PIPA	Pulse Integrating Pendulous Accelerometers
PLM	LM Pitch Angle
PRO	Proceed
PROG	Program
PROP	Propellant
R	Range
RCS	Reaction Control System
RDOT	Range Rate
REFSMMAT	Reference Stable Member Matrix
RHC	Rotation Hand Controller
RR	Rendezvous Radar
SHFT	Shaft
SC	Spacecraft
SEP	Separation
SM	Service Module
SXT	Sextant
SYNC	Synchronize
TFI	Time From Ignition
THC	Translation Hand Controller
THETA	Angle Between SC +X Axis and Local Horizontal
TIGN	Time of Ignition
TLM	Telemetry
TPF	Terminal Phase Finalization
TPI	Terminal Phase Initiation
TRUN	Trunnion
TTCA	Translation Thrust Control Assembly
VG	Velocity to be Gained
VHF	Very High Frequency
(XX:XX)	Indicates GET From Liftoff in Hours:Minutes
(XXX:XX XX)	Indicates GET From Liftoff in Hours:Minutes:Seconds
(XXX,XXX/XXX,XXX)	(Ordeal/Inertial) Angles (Roll, Pitch, Yaw)
(XX,XX,XX)	Local Vertical DV's

Tracking Stations

ANG	Antigua Near Space Support Station
BDA	Bermuda Near Space Support Station
CRO	Carnarvon Near Space Support Station
CYI	Canary Near Space Support Station
GYM	Guaymas Near Space Support Station
HSK	Honeysuckle Deep Space Support Station
HTV	Huntsville Near Space Support Station
MAD	Madrid Deep Space Support Station
MER	Mercury Near Space Support Ship
MIL	MILA Near Space Support Station
RED	Redstone Near Space Support Ship
TEX	Corpus Christi Near Space Support Station
VAN	Vanguard Near Space Support Ship

1.0 PURPOSE

This document contains the primary procedures for the LM-5 active rendezvous with CSM-107 spacecraft, in accordance with Detailed Test Objective Section 4.0 defined in Reference 7.1.

The purpose of the LM Rendezvous Procedures document is to provide a single source of procedures information for use in flight planning, in crew training, and in preparing onboard data.

This is a control document, subject to review by all elements of the Apollo Program and to approval by the Procedures Configuration Control Board. Comments should be directed to Mr Stephen P. Grega, Flight Procedures Branch, Flight Crew Support Division, Extension 5348.

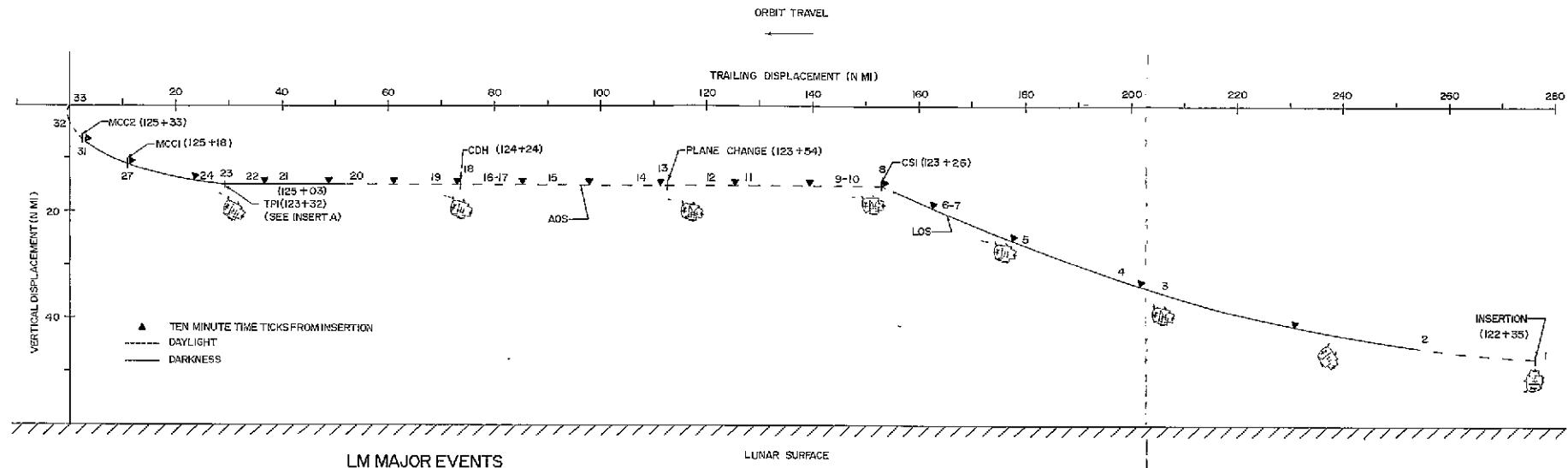
Beyond G mission planning, no additional techniques and trajectory information specific to Mission H have been made available as of this date. Therefore, since Missions G and H are expected to be similar in the area of rendezvous guidance and navigation procedures, the G mission final "LM Rendezvous Procedures Document" will serve as an H mission preliminary "LM Rendezvous and Procedures Document". After reviewal by all elements of the Apollo Program and approval by the Procedures Configuration Control Board, this document will then evolve into the control document for the Mission H crew procedures.

2.0 INTRODUCTION

The LM-5/CSM-107 lunar orbit rendezvous exercise will begin during the twenty-fourth CSM revolution with insertion at 122 35:25.5 and end at approximately 126:00:00 with post rendezvous station keeping. The LM procedures during this period are divided into segments of major activities which are discussed in detail in Section 3.0.

A nominal LM-5/CSM-107 mission profile is contained in Figure (2-1). This figure shows the locations in time and local horizontal position of the most significant nominal mission events. Trajectory data used to generate the mission profile and timeline for procedures development was obtained from Reference 7.2. The rendezvous navigation update schedule assumed in the procedures is defined in Reference 7.3.

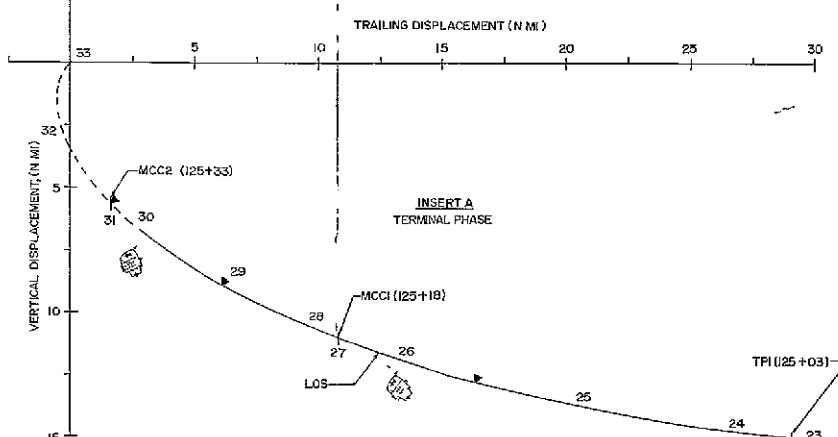
A history of the LM body attitudes during the rendezvous accompanies the major events discussion and is also presented in Figures (2-2) and (2-3). Each figure illustrates the body attitudes with respect to the moon, sun, and earth and indicates FDAO roll, pitch, and yaw gimbal angles and the ordeal pitch angle for significant events during each lunar orbit.



1 122+35	INSERTION	17 124+17	AGS UPDATE AND ALIGN (P47)
2 122+40	IMU REFSMMAT ALIGN (P52)	18 124+24	CDH THRUSTING (P41)
3 122+54	AGS UPDATE AND ALIGN (R47)	19 124+26	TPI TARGETING (P34 & AGS)
4 122+55	CSI TARGETING (P32 & AGS)	20 124+26—124+51	RR NAVIGATION (P20, 18 MARKS)
5 122+54—123+14	RR NAVIGATION (P20, 18 MARKS)	21 124+53	AGS UPDATE AND ALIGN (R47)
6 123+18	COMPUTE CSI BACKUP	22 125+00	COMPUTE TPI BACKUP
7 123+19	AGS UPDATE AND ALIGN (R47)	23 125+03	TPI THRUSTING (P41)
8 123+26	CSI THRUSTING (P41)	24 125+05	MCC1 TARGETING (P35)
9 123+30	SET W MATRIX [2000,2,5]	25 125+05—125+16	RR NAVIGATION (P20, 9 MARKS)
10 123+31	CDH TARGETING (P33 & AGS)	26 125+16	COMPUTE MCC1 THRUSTING (P41)
11 123+31—123+46	RR NAVIGATION (P20, 14 MARKS)	27 125+18	MCC1 THRUSTING (P41)
12 123+47	EXT ΔV FOR PLANE CHANGE (P30)	28 125+20	MCC2 TARGETING (P35)
13 123+54	RCS PLANE CHANGE THRUSTING (P41)	29 125+20—125+31	RR NAVIGATION (P20, 9 MARKS)
14 123+56	CDH TARGETING (P33 & AGS)	30 125+31	COMPUTE MCC2 BACKUP
15 123+56—124+12	RR NAVIGATION (P20, 13 MARKS)	31 125+33	MCC2 THRUSTING (P41)
16 124+15	COMPUTE CDH BACKUP	32 125+48	BRAKING (P47)
		33 126+00	STATION KEEPING

FIGURE 2-1  
G MISSION RENDEZVOUS  
(CSM-CENTERED RELATIVE MOTION)

FOLDOUT FRAME



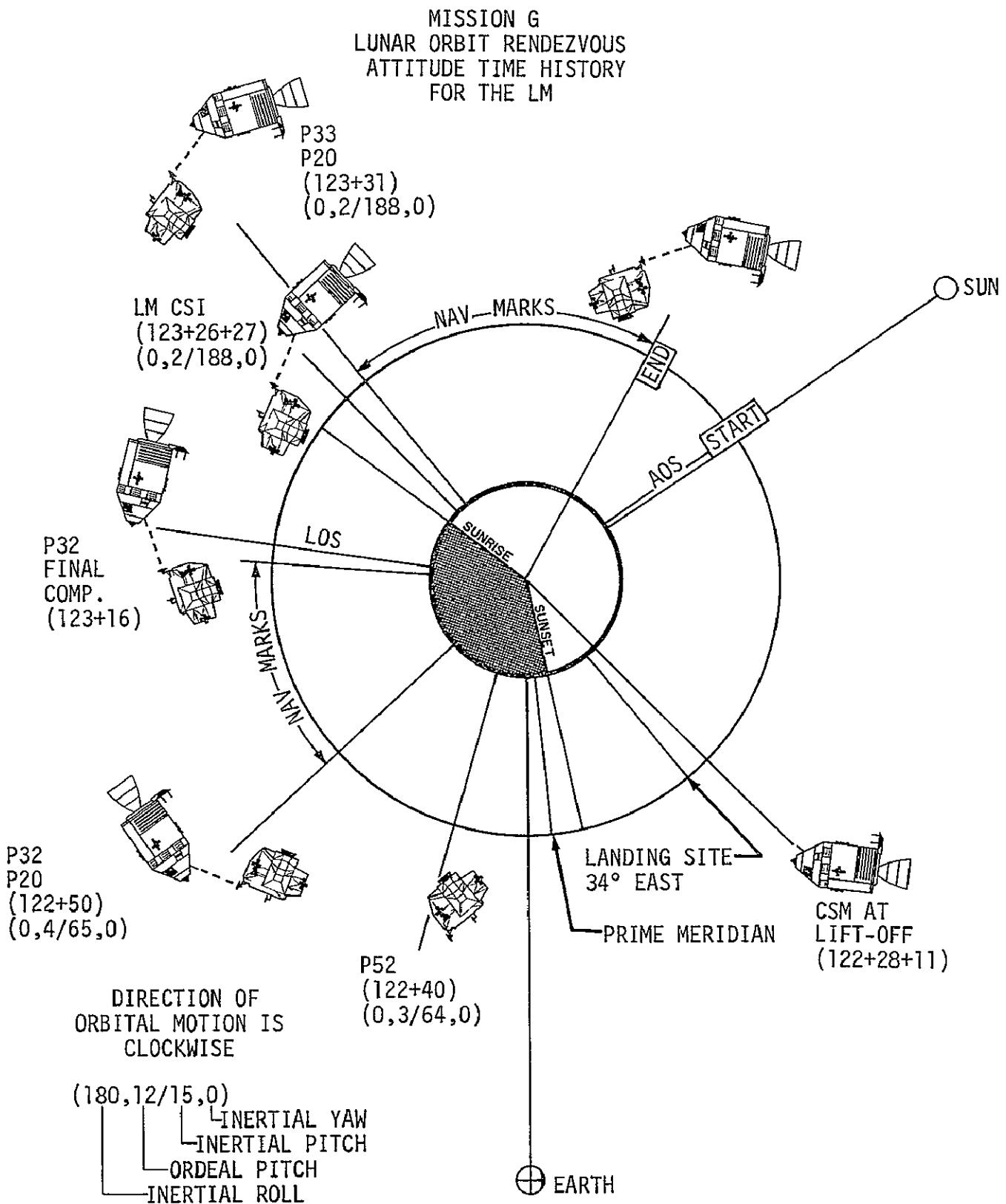


FIGURE 2-2

MISSION G  
LUNAR ORBIT RENDEZVOUS  
ATTITUDE TIME HISTORY  
FOR THE LM

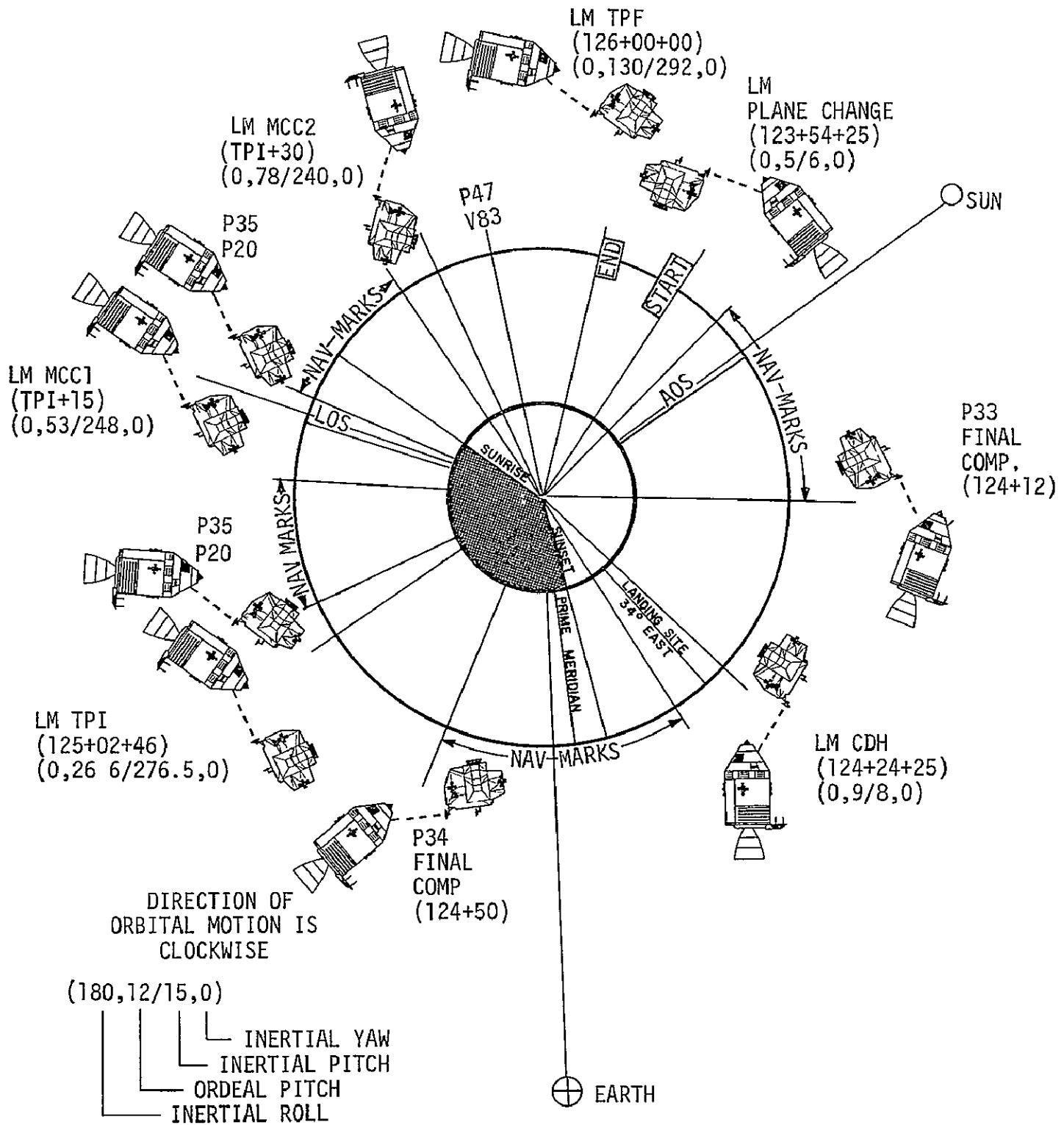


FIGURE 2-3

3.0 DISCUSSION OF MAJOR EVENTS3.1 Concentric Sequence Initiation

Immediately following Insertion the LM navigated vector will be downlinked to Mission Control Center-Houston, this state vector is then uplinked to the CSM. The LM commander will perform a Program 52 fine alignment to REFSMMAT at approximately 122.40.

The digital autopilot is configured to wide deadband, Program P20 is called, and an automatic pitch maneuver is initiated to the preferred track attitude. At 122 55, CSI targeting parameters are loaded in the PGNS and AGS. A period of rendezvous radar marking will then take place from 122.57 to 123.15. Upon completion of tracking, the LM will obtain an out-of-plane velocity from the CSM to be vectored with the in-plane component of the CSI delta V. The CSM's estimate of out-of-plane velocity is alleged to be better than the LM's estimate. (Reference 7.4.)

At approximately 123 20 the LM will voice its CSI solution to the CSM; the AGS is updated, aligned, and targeted in external delta-V with the CDH solution to be excuted. Program P41 will then be called and the automatic pitch maneuver bypassed since the burn is to be executed with Z-Axis thrusting. The CSI burn occurs nominally at 123 26 27. After verification of the LM CSI burn, the CSM will incorporate the LM burn parameters in its CMC LM state vector

3.2 Constant Delta Height and Plane Change

At 123:30 approximately three minutes after the LM CSI burn, the W matrix will be set to (2000,2,5) and the LM targeting paramters are loaded in both the PGNS and AGS for the CDH maneuver. Out-of-plane velocity will then be computed and voiced to the LM from the CSM. Program P30 is loaded with the targeting parameters for the LM Plane Change maneuver which will be accomplished in Program P41 30 minutes prior to the CDH maneuver. Following the burn, the target delta V parameters are voice to the CSM and incorporated into the CMC LM state vector. At 123:56 the CDH targeting parameters are loaded in both the PGNS and AGS for 124.24:25. A CDH chart solution is calculated 7 minutes prior to TIG. The AGS is updated, aligned, and targeted in external delta V with the CDH solution to be executed. The RCS Thrust Program, P41, is called at 124:18 and the burn attitude is bypassed since the LM executes CDH by thrusting out individual components of delta V while maintaining radar lock-on.

3 3 Terminal Phase Initiation

Radar tracking is initiated after the CDH maneuver and updating of the LGC continued. At 124:26, approximately two minutes after the LM CDH burn, targeting parameters are loaded in both the PGNS and AGS for the TPI maneuver which will be targeted for a node at intercept. Backup measurements are made at the appropriate times. The AGS is updated, aligned, and targeted in external delta V with the TPI solution to be executed. The RCS Thrust Program, P41, is called at 124:56 and the burn attitude is bypassed since the LM is burning along the plus Z-axis with the thrust direction established with the COAS.

3 4 Midcourse Corrections

Immediately after TPI the Midcourse Prethrust Program, P35, will be called and radar tracking reinitiated. Backup measurements are made and midcourse corrections chart solutions obtained. The MCC1 burn will nominally be performed at TPI plus 15 minutes (125:17:46). If MCC1 is to be excuted, Program P41 will be called and the burn attitude maneuver is bypassed since the plus Z-axis will remain boresighted at the CSM until docking.

Following MCC1, Program P35 will be called and targeted for MCC2 at TPI plus 30 minutes (125:32.46) and executed identically to MCC1.

3 5 Braking

In preparation for braking, the DAP is configured in narrow deadband. Program P20 is terminated at 18,000 feet but the radar is kept in auto track to provide radar range and range rate displayed on the tapemeter. Average "G" is called by Program P47 prior to braking, and attitude hold is selected. The braking schedule is followed as contained in Section 4.

4.0 NOMINAL MISSION PROCEDURES

The procedures included in this section do not specify which crewmember, Commander or LM Pilot, performs each task. These need not be specified since the following basic rules define which crewmember performs each task.

4.1 Commander Tasks

- 1 All attitude changes, whether manual or automatic, will be accomplished by the CDR.
- 2 The operation of the DSKY during thrust programs (P40, P41, P42, P47) will be done by the CDR. Actual manipulation of the TTCA need not be always done by the CDR, but in most cases will be.
3. Operation of the rendezvous and landing radar will be done by the CDR.
- 4 The CDR will operate all other systems accessible to only his crew station.

4.2 LM Pilot Tasks

- 1 AGS operation.
2. DSKY operation will be done by the LMP except when keyboard entries affect the control of spacecraft attitude or thrusting.
3. Backup data logging and chart calculations.
- 4 All logging of maneuver solutions and systems performance.
- 5 The LMP will operate all other systems accessible to only his crew station.

These general rules are guidelines only, and may be deviated from by the LM crew if they develop more efficient Task assignments.

4.3 ABBREVIATIONS

The abbreviations used herein are consistent with those in the AOH. However, in order to condense and simplify the procedures so that they are representative of onboard data, a number of additional shorthand conventions have been used. To allow the unfamiliar reader to understand the procedures contained in this section, the following explanations are included

1. AGS - A single asterisk is employed to denote those procedures involving operations to be performed on the DEDA. The three number group following an asterisk specifies DEDA address. An "R" following the address group indicates the address is to be read out. A five digit group behind the "R" indicates a nominal or expected display. If the three number address group is followed by a plus or minus sign, a data load is indicated. When a single digit follows the plus, addition of four zeros behind this digit is assumed. Once used, these conventions are easily handled and save considerable space.

2. PGNS - The verb-noun addresses in the PGNS are indicated to the left of the procedures column. An "F" is used to indicate a flashing display, or absence of an "F" a static display. To the right of the verb-noun, on the same line, are the contents of the three data registers. If numeric quantities appear, the DSKY should be correspondingly made to agree by executing a V21, V22, or V23 and performing a data load. The procedure of blanking and loading registers is not included since it is repeated often and is highly familiar to the crews.

EXAMPLE F 06 33 125.02 46 TIG OF TPI

Expanded, this means; Load flashing verb 06 noun 33 with the quantity 125 hours, 2 minutes and 46 seconds, the time of ignition of the RCS engine TPI maneuver. Procedurally, this is done by keying verb 25 enter, loading +000125 in register 1, +00002 in register 2, and +04600 in register 3.

If numeric quantities do not follow the verb-noun, it is indicated that a computed or information quantity is being displayed via the DSKY. If nominal values are listed for possible comparison, they appear in parentheses.

EXAMPLE F 06 42 HA HP V

Expanded, this means Flashing verb 06 noun 42 has computed height of apogee in register 1, height of perigee in register 2, and velocity to be gained in register 3.

3. TIME - The numbers in the left hand column preceded by a plus or minus indicate "event timer" time referenced to an event. Minus indicates event timer counting down to a future event. Periodically, this same column contains the nominal mission ground elapsed times in parentheses. Asterisks following the event time indicate a time critical event which should be accomplished to the nearest second.

4 4 Nominal PGNS Procedures

## ASSUMPTIONS

\*\*\*\*\*  
 \* AFTER INSERTION \*  
 \*\*\*\*\*

- 1 INSERTION COMPLETE (RESIDUALS NULLED)
- 2 VERIFY V82E (ORBIT)
- 3 RNDZ RDR LOCKED ON IN MODE II
- 4 LGC AT FLASHING VERB 37
- 5 PROPER INITIAL W MATRIX LOADED IN LGC

GUID CONT-PGNS  
 ATTITUDE MON-PGNS(CDR)  
 RATE ERR MON-CMPTR(LMP)  
 ATTITUDE MON-AGS(LMP)  
 SHFT/TRUN-+5 DEG  
 X-POINTER-HI MULT  
 RADAR TEST SW-OFF  
 ENG ARM-OFF  
 X-TRANSL-2JETS  
 DEAD BAND-MIN  
 PGNS MODE CONT-AUTO  
 THROT/JET-JET  
 CB/PGNS IMU OPR-CLOSE  
 R/R MODE-LGC  
 TTCA/TRANSL-ENABLE(CDR)

\*\*\*\*\*  
 \* BAL CPL-ON \*  
 \* MODE SEL-AGS \*  
 \* RNG/ALT MON-RNG/RNGRT \*  
 \* RATE ERR MON-RNDZ RDR(CDR) \*  
 \* ATT CONT-PULSE(3) \*  
 \* AGS MODE CONT-ATT HOLD \*  
 \* ACA/4JET-DISABLE(CDR) \*  
 \* TTCA/TRANSL-DISABLE(LMP) \*  
 \* ACA/4JET-DISABLE(LMP) \*  
 \*\*\*\*\*

KEY 00E (POO)  
 (DOWNLINK LM STATE VECTOR)

KEY V63E (RR SELF TEST)

F 04 12 00004 00001  
PRO  
F 16 72 TRUN SHFT  
PRO  
F 16 78 R RDOT  
VERIFY TAPE METER WITH DSKY

\*\*\*\*\*  
+2 \* VERIFY INSERTION VELOCITY \*  
\* WITH RNDZ RDR, WHERE \*  
\* RANGE = 264.2 N.M. \*  
\* RANGE RATE = -425.1 FPS \*  
\*\*\*\*\*

SUIT/CABIN PRESS - NOR  
GLYCOL TEMP/PRESS - NOR  
O2/H2O QUANTITY - NOMINAL

KEY V34E (TERMINATE)  
KEY V48E (LOAD DAP)  
F 01 46 11002  
PRO  
F 06 47 LMWT CMWT  
PRO

KEY V56E (KILL AUTO TRACK ENABLE)

+3 KEY V41N72E (COARSE ALIGN R/R)  
F 21 73 +000.00TRUN -077.00SHFT  
F 04 12 00006 00002 (CONTINUOUS DESIGNATE)  
PRO  
41 (COARSE ALIGN VERB)

KEY V16N72E  
16 72 MONITOR TRUN/SHFT ANGLES

CB PGNS RNDZ RDR-OPEN  
CB AC BUS A RNDZ RDR-OPEN  
KEY V44E (TERMINATE RR DESIGNATE)

KEY V83E  
F 16 54 R RDOT THETA  
SET ORDEAL  
PRO

CB HEATER AOT-CLOSE  
CB AOT LAMP-CLOSE

+5 KEY V37E52E (IMU REALIGN)

\*410+1 CSI ROUTINE  
 \*373+0206.5 CSI TIME  
 \*275+0302.8 TPI TIME  
 \*605+00777 COTANGENT OF LOS. TO CSM  
 \*416+1 (ONE-HALF ORBITAL PERIOD)  
 \*411+0 RCS ENGINES  
 \*623+0 Z-AXIS PARALLEL TO CSM ORBIT  
 \*310R TIME TO CSI

\*267R \_\_\_\_\_ VELOCITY TO BE GAINED CSI

\*402R \_\_\_\_\_ DELTA H

\*371R \_\_\_\_\_ DELTA V CDH

\*372R \_\_\_\_\_ DELTA T CSI/CDH

F 04 06 00001 00003 (REFSMMAT)

PRO

F 50 25 00015 SELECT 1ST STAR  
ENTR

F 01 70 002DE LOAD 1ST STAR  
PRO

PGNS MODE CONT-AUTO

F 50 18 FDAI ANGLES (REQUEST MNVR)  
PRO (AUTO MNVR)

06 18 FDAI ANGLES (AUTO MNVR)

F 50 18 FDAI ANGLES (REQUEST MNVR)  
ENTR (BYPASS MNVR)

F 01 71 002DE

KEY V76E

PGNS MODE CONT-ATT HOLD

PRO

+10 F 54 71 MARK 1ST STAR  
PRO

F 01 70 002DE LOAD 2ND STAR  
PRO

PGNS MODE CONT-AUTO

F 50 18 FDAI ANGLES (REQUEST MNVR)  
PRO (AUTO MNVR)

06 18 FDAI ANGLES (AUTO MNVR)

F 50 18 FDAI ANGLES (REQUEST MNVR)  
ENTR (BYPASS MNVR)

F 01 71 002DE

PGNS MODE CONT-ATT HOLD

PRO

+14 F 54 71 MARK 2ND STAR  
PRO  
F 06 05 STAR ANGLE DIFFERENCE  
PRO  
F 06 93 GYRO TORQUE ANGLES  
PRO  
F 50 25 00014

PRO (CHECK ALIGNMENT IF TIME WILL PERMIT)

F 50 25 00015 SELECT 3RD STAR  
ENTR  
F 01 70 002DE LOAD 3RD STAR  
PRO

PGNS MODE CONT-AUTO  
F 50 18 FDAI ANGLES (REQUEST MNVR)  
PRO (AUTO MNVR)  
06 18 FDAI ANGLES (AUTO MNVR)  
F 50 18 FDAI ANGLES (REQUEST MNVR)  
ENTR (BYPASS MNVR)

VERIFY ALIGNMENT VIA AOT

KEY V34E (TERMINATE)

F 37 BB

V48E (LOAD DAP)  
F 01 46 11012  
PRO

+17 F 06 47 LMWT CMWT  
PRO

CB AOT LAMP-OPEN

-35 RESET ET (310R)

(122:53) CB AC BUS A RNDZ RDR-CLOSE  
(WAIT 30 SEC)  
CB PGNS RNDZ RDR-CLOSE

RR MODE-LGC  
KEY V37E20E (ACQUIRE RADAR)

PGNS MODE CONT-AUTO  
F 50 18 FDAI ANGLES (REQUEST MNVR)  
PRO (AUTO MNVR)  
06 18 FDAI ANGLES (AUTO MNVR)  
AGS MODE CONT-AUTO  
RATE/ERR MON-CMPTR (LMP)  
F 50 18 FDAI ANGLES (REQUEST MNVR)

ENTR (BYPASS MNVR)  
 NO TRK LT-OFF  
 F 50 72 RR TRUN/SHFT  
 VERIFY R/R LOCK ON  
 PRO  
 DSKY BLANKS

KEY V93E (BEFORE 1ST MARK REINIT W. MATRIX)  
 KEY V80E (UPDATE LM SV)

KEY V47E (INITIALIZE AGS)  
 F 06 16 GET OF AGS 0 TIME (120:00:00)  
 TELEMETRY PCM SW-HI  
 \*414+1 UPDATE AGS .  
 PRO  
 \*414R (00000 COMPLETE)  
 F 50 16 DOWNLINK COMPLETE  
 PRO (UPDATE COMPLETE)  
 \*400+3 AGS ALIGN -  
 \*400R (00000 COMPLETE)  
 \*507+0 Z-AXIS BORESIGHT  
 \*400+2 ACQUISITION STEERING

\*\*\*\*\*  
 \* CSI \*  
 \*\*\*\*\*

(122&55) KEY V37E32E (CSI)

F 06 11 : : LOAD TIG CSI FROM DATA PAD  
 (NOM 123:26:27)  
 PRO  
 F 06 55 +00001N +026.60E +130.00 OMEGAT  
 (CDH 180 DEG AFTER CSI)  
 PRO

F 06 37 : : LOAD TIG TPI FROM DATA PAD  
 (NOM 125:02:46)  
 PRO

F 16 45 MKS TFI -00001

-30\* COPY RDOT FOR 1ST CSI BU

\*310R TIME TO CSI

\*267R \_\_\_\_\_ VELOCITY TO BE GAINED CSI

\*402R \_\_\_\_\_ DELTA H

\*371R \_\_\_\_\_ DELTA V CDH

\*372R \_\_\_\_\_ DELTA T CSI/CDH

KEY V32E (MARKS=5)

F 06 75 DELTA H DELTA T CSI/CDH DELTA T CDH/TPI

PRO

F 06 81 DELTA V'S-LV (CSI)

(NOM 50.1 DELTA VX)

CSI(N81)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

PRO

F 06 82 DELTA V'S-LV (CDH)

(NOM 6.0 DELTA VX)

CDH(82)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

PRO

F 16 45 MKS TFI -00001

-22 KEY V32E (MARKS=10)

F 06 75 DELTA H DELTA TCSI/CDH DELTA TCDH/TPI

PRO

F 06 81 DELTA V'S-LV (CSI)

CSI(N81)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

PRO

F 06 82 DELTA V'S-LV(CDH)

CDH(82)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

-20\* COPY RDOT FOR 2ND CSI BU

PRO  
F 16 45 MKS TFI -00001

TEMP/PRESS MON SEL - FUEL MAN  
A + B PRESS 181 +-3 PSIG  
TEMP/PRESS MON SEL - PRPLNT  
A + B TEMP IND 40DEG TO 100DEG F  
A + B PRESS IND 181 +-3 PSIA  
TEMP/PRESS MON SEL - HEL

A + B PRESS IND \_\_\_\_\_ PSIA

A QUANTITY IND \_\_\_\_\_ PERCENT REMAINING

B QUANTITY IND \_\_\_\_\_ PERCENT REMAINING  
TEMP MON SEL RCS QUAD 1, 4, 2, 3 (120 - 190 DEG F )

(LOS OF MSFN )  
S-BAND - SLEW TO P = +90, Y = 0  
S-BAND - FWD (OR AFT)  
S-BAND - DN VOICE BU  
BIOMED OFF

-12 PRO (FINAL COMP)

\*267R \_\_\_\_\_ VELOCITY TO BE GAINED CSI  
\*402R \_\_\_\_\_ DELTA H  
\*372R \_\_\_\_\_ TIME CSI/CDH  
\*371R \_\_\_\_\_ PREDICTED VEL TO BE GAINED CDH  
\*263R \_\_\_\_\_ OUT-OF-PLANE

F 06 75 DELTA H DELTA TCSI/CDH DELTA TCDH/TPI  
PRO

F 06 81 DELTA V'S-LV (CSI)

KEY V90E (OUT-OF-PLANE)

F 06 16 : : LOAD TIG CSI FROM DATA PAD  
(NOM 123:26:27)  
PRO

F 06 90 Y YDOT PSI  
N90 CSM

YDOT \_\_\_\_\_

PRO

F 06 81 DELTA V\*S-LV(CSI)

COPY YDOT FROM CSM AND LOAD NEGATIVE IN R2  
CSI(N81)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

PRO

F 06 82 DELTA V\*S-LV (CDH)

CDH(82)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

PRO

F 16 45 MKS TFI MGA

-10\* COPY RANGE AND RDOT FOR 3RD CSI BU

COMPUTE CSI BU

-7 KEY V47E (INITIALIZE AGS)

F 06 16 GET OF AGS 0 TIME (120.00.00)

TELEMETRY PCM SW-HI

\*414+1 UPDATE AGS

PRO

\*414R (00000 COMPLETE)

F 50 16 DOWNLINK COMPLETE

PRO (UPDATE COMPLETE)

F 16 45 MKS TFI MGA

(TRANSMIT CSI DATA TO CSM)

VHF A XMTR - VOICE

VHF B XMTR - DATA

VHF B RCVR - OFF

TELEMETRY PCM - LO

KEY V83E

F 16 54 R RDOT THETA

SET ORDEAL

PRO

\*317R \_\_\_\_\_ RANGE

\*440R \_\_\_\_\_ RANGE RATE

\*400+3 AGS ALIGN  
\*400R (00000 COMPLETE)

\*277R                    THETA

PRO

F 37 BB

-6           KEY 41E (RCS THRUSTING)  
F 50 18 FDAI ANGLES (REQUEST MNVR)  
ENTR (BYPASS MNVR)

16 85 DELTA V\*S-BODY

KEY V06N86E DELTA V\*S-LV  
F 06 86 DELTA V\*S-LV

AGS(N86)    CHART

DELTA VX(LV)                                      

DELTA VY(LV)                                      

DELTA VZ(LV)                                      

KEY RELEASE

\*507+1 Z-AXIS DELTA V  
\*400+2 Z-AXIS STEERING  
\*410+5 EXTERNAL DELTA V  
\*450 451 452 LOAD  
\*407+0 ROTATE EXTERNAL DELTA V RF

TRANSMIT TGT DELTA V\*S TO CSM AND SYNC COUNTDOWN

16 85 DELTA V\*S-BODY

\*502R                    DELTA VZ (NOM 50.1)

-35    DSKY BLANKS

KEY V77E (ATT HOLD)  
PGNS MODE CONT-ATT HOLD  
AGS MODE CONT-ATT HOLD  
\*407+1 FREEZE EXTERNAL DELTA V

F 16 85 DELTA V\*S-BODY

(123°26'27") BURN +Z

NULL DELTA V\*S

VERIFY AGS RESIDUALS

\*500R 501R 502R  
DELTA VX \_\_\_\_ DELTA VY \_\_\_\_ DELTA VZ \_\_\_\_

VERIFY PGNS RESIDUALS

N85

DELTA VX(LM) \_\_\_\_\_

DELTA VY(LM) \_\_\_\_\_

DELTA VZ(LM) \_\_\_\_\_

PRO

F 37 BB

\*\*\*\*\*  
\* CDH \*  
\*\*\*\*\*

KEY V76E (PGNS PULSE)

PGNS MODE CONT-AUTO

KEY V67E (RESET W MATRIX)

F 06 99 +02000E +00020E +00005E  
PRO

\*507+0 Z-AXIS BORESIGHT

(VHF RANGING WITH CSM)  
VHF A XMTR - VOICE/RNG  
VHF B XMTR - OFF  
VHF B RCVR - ON  
TELEMETRY PCM - HI

(123:31) KEY V37E33E (CDH PRE-THRUST)

\*410+2 CDH ROUTINE

F 06 13 : : TIG CDH (NOM 124:24:25)  
-120 :00 :00 (AGS 0 TIME)

(CONVERT TO MINUTES)

\*411+0 THRUST ALONG Z-BODY AXIS  
ENTER IF YOU DON'T TARGET CSI\*373R \_\_\_\_\_ (NOM 264.4)  
\*310R TIME TO CDH

\*423R \_\_\_\_\_ DESIRED ALT RATE AT CDH

\*402R \_\_\_\_\_ DELTA H

\*450R \_\_\_\_\_ DELTA VX

\*452R \_\_\_\_\_ DELTA VZ

\*623+0 Z-AXIS PARALLEL TO CSM PLANE

PRO  
F 16 45 MKS TFI -00001

-52           RESET ET

KEY V93E (REINIT W MATRIX WHEN MARKS = 4)

KEY V32E (MARKS = 3)

F 06 75 DELTA HCDH DELTA TCDH/TPI DELTA TTPI  
PRO

F 06 81 DELTA VCDH-LV                                   CDH(N81)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

PRO  
F 16 45 MKS TFI -00001

-47           KEY V90E (OUT-OF-PLANE)

\_\_\_\_\_ : : TIG CDH (NOM 124:24:25)  
-30:00 TIME FROM CDH

F 06 16       : : LOAD TIG PLANE CHANGE  
(NOM 123:54:25)

PRO  
F 06 90 Y YDOT PSI  
N90   CSM

YDOT \_\_\_\_\_  
PRO

-42           KEY V32E (MARKS=10)

F 06 75 DELTA HCDH DELTA TCDH/TPI DELTA TTPI/TPI  
PRO

F 06 81 DELTA VCDH-LV

4-14

CDH(N81)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

PRO

F 16 45 MKS TFI -00001  
KEY V34E (TERMINATE)

F 37 BB

\*\*\*\*\*  
\* PLANE CHANGE \*  
\*\*\*\*\*

KEY 30E (EXTERNAL DELTA V)

\_\_\_\_\_ : : TIG CDH (NOM 124:24:25)  
-30:00 TIME FROM CDH

F 06 33 \_\_\_\_\_ : : LOAD TIG PLANE CHANGE  
(NOM 123:54:25)

PRO

F 06 81 DELTA V\*S-LV (NOM 0.0)

-36\* COPY YDOT FOR 1ST CDH BU

KEY V90E (OUT-OF-PLANE)

F 06 16 \_\_\_\_\_ : : LOAD TIG PLANE CHANGE  
(NOM 123:54:25)

PRO

F 06 90 Y YDOT PSI  
N90 CSM

YDOT \_\_\_\_\_  
PRO

F 06 81 DELTA V\*S-LV(NOM 0.0)

COPY YDOT AND LOAD NEGATIVE IN R2  
PRO

F 06 42 HA HP DELTA V

PRO

F 16 45 MKS TFI MGA

PRO

F 37 BB

KEY 41E (RCS THRUSTING)

F 50 18 FDAI ANGLES (REQUEST MNVR)  
ENTR (BYPASS MNVR)

\*410+5 EXTERNAL DELTA V  
\*450+0 451-YDOT 452+0  
(LOAD CSM -YDOT)  
\*407+0 ROTATE EXTERNAL DELTA V RF

16 85 DELTA V'S-BODY

-30:35 DSKY BLANKS

KEY V77E (ATT HOLD)  
PGNS MODE CONT-ATT HOLD  
AGS MODE CONT-ATT HOLD  
\*407+1 FREEZE EXTERNAL DELTA V

F 16 85 DELTA Y'S-BODY

-30 NULL DELTA V'S

PRO

F 37 BB

KEY V76E (PGNS PULSE)  
PGNS MODE CONT-AUTO  
\*507+0 Z-AXIS BORESIGHT

(123:56) KEY 33E (CDH PRE-THRUST)

\*410+2 CDH ROUTINE

F 06 13 : : TIG CDH (NOM 124:24:25)  
-120 :00 :00 (AGS 0 TIME)

(CONVERT TO MINUTES)

\*373R \_\_\_\_\_ (NOM 264.4)  
\*310R TIME TO CDH

\*423R \_\_\_\_\_ DESIRED ALT RATE AT CDH

(AOS OF MSFN)  
S-BAND - SLEW TO P = \_\_\_\_\_, Y = \_\_\_\_\_  
ACQUIRE HGA LOCK ON \_\_\_\_\_  
VOICE - BU/VOICE -- VOICE  
VERIFY VOICE, TLM  
BIOMED - LEFT (OR RIGHT)  
\*402R \_\_\_\_\_ DELTA H

\*450R \_\_\_\_\_ DELTA VX

\*452R \_\_\_\_\_ DELTA VZ

PRO  
F 16 45 MKS TFI -00001

-25 KEY V93E (REINIT W MATRIX WHEN MARKS = 4)

-23\* COPY RDOT FOR 2ND CDH BU

KEY V32E (MARKS = 5)

TEMP/PRESS MON SEL - FUEL MAN

A + B PRESS 181 +-3 PSIG

TEMP/PRESS MON SEL - PRPLNT

A + B TEMP IND 40DEG TO 100DEG F

A + B PRESS IND 181 +-3 PSIA

TEMP/PRESS MON SEL - HEL

A + B PRESS IND \_\_\_\_\_ PSIA

A QUANTITY IND \_\_\_\_\_ PERCENT REMAINING

B QUANTITY IND \_\_\_\_\_ PERCENT REMAINING

TEMP MON SEL RCS QUAD 1, 4, 2, 3 (120 - 190 DEG F)

SUIT/CABIN PRESS - NOR

GLYCOL TEMP/PRESS - NOR

O2/H2O QUANTITY - NOMINAL

F 06 75 DELTA HCDH DELTA TCDH/TPI DELTA TTPI/TPI  
PRO

F 06 81 DELTA VCDH-LV

CDH(N81)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

PRO

F 16 45 MKS TFI -00001

-12 PRO (FINAL COMP)

F 06 75 DELTA HCDH DELTA TCDH/TPI DELTA TTPI/TPI  
PRO

F 06 81 DELTA VCDH-LV

KEY V90E (OUT-OF-PLANE)

F 06 16 : : LOAD TIG CDH FROM DATA PAD  
(NOM 124:24:25)

PRO

F 06 90 Y YDOT PSI  
N90

CSM

YDOT \_\_\_\_\_  
PRO \_\_\_\_\_

COPY YDOT FROM CSM AND LOAD NEGATIVE IN R2  
CDH(N81)

DELTA VX(LV) \_\_\_\_\_

DELTA VY(LV) \_\_\_\_\_

DELTA VZ(LV) \_\_\_\_\_

PRO

F 16 45 MKS TFI MGA

-10\* COPY RDOT FOR 3RD CDH BU

COMPUTE CDH BU

TRANSMIT TGT DELTA V'S TO CSM AND SYNC COUNTDOWN

-7 KEY V47E (INITIALIZE AGS)

F 06 16 GET OF AGS O TIME (120:00:00)  
TELEMETRY PCM SW-HI

\*414+1 UPDATE AGS

PRO

\*414R (00000 COMPLETE)

F 50 16 DOWNLINK COMPLETE  
PRO (UPDATE COMPLETE)

F 16 45 MKS TFI MGA

KEY V83E

F 16 54 R RDOT THETA  
SET ORDEAL  
PRO

\*317R \_\_\_\_\_ RANGE

\*440R \_\_\_\_\_ RANGE RATE

\*400+3 AGS ALIGN

\*400R (00000 COMPLETE)

\*277R \_\_\_\_\_ THETA

PRO

F 3.7 BB

-6           KEY 41E (RCS THRUSTING)  
 F 50 18 FDAO ANGLES (REQUEST MNVR)  
 ENTR (BYPASS MNVR)

16 85 DELTA V'S-BODY

KEY V06N86E  
 F 06 86 DELTA V'S-LV

AGS(N86)    CHART

DELTA VX(LV)    \_\_\_\_\_

DELTA VY(LV)    \_\_\_\_\_

DELTA VZ(LV)    \_\_\_\_\_

KEY RELEASE

\*507+1 Z-AXIS DELTA V  
 \*400+0 ATT HOLD  
 \*410+5 EXTERNAL DELTA V  
 \*450    451    452 LOAD  
 \*407+0 ROTATE EXTERNAL DELTA V RF

16 85 DELTA V'S-BODY

\*500R \_\_\_\_\_ DELTA VX (NOM -0.3)

\*502R \_\_\_\_\_ DELTA VZ (NOM 6.0)

-:35 DSKY BLANKS

KEY V77E (ATT HOLD)  
 PGNS MODE CONT-ATT HOLD  
 AGS MODE CONT-ATT HOLD  
 \*407+1 FREEZE EXTERNAL DELTA V

F 16 85 DELTA V'S BODY

(124:24:25) NULL DELTA V'S

VERIFY AGS RESIDUALS

\*500R       501R       502R  
 DELTA VX \_\_\_\_\_ DELTA VY \_\_\_\_\_ DELTA VZ \_\_\_\_\_

VERIFY PGNS RESIDUALS

N85

DELTA VX(LM)    \_\_\_\_\_

DELTA VY(LM) \_\_\_\_\_

DELTA VZ(LM) \_\_\_\_\_

PRO

F 37 8B

\*\*\*\*\*  
 \* TPI \*  
 \*\*\*\*\*

KEY V76E (PGNS PULSE)  
 PGNS MODE CONT-AUTO  
 \*507+0 Z-AXIS BORESIGHT

(124:26) KEY 34E (TPI)

-36 RESET ET

F 06 37 : : LOAD TIG TPI FROM DATA PAD  
 (NOM 125:02:46)

PRO

F 06 55 +00000 +026.60E +130.00 OMEGAT  
 PRO

F 16 45 MARKS TFI -00001

-32 KEY V93E (REINIT W MATRIX WHEN MARKS = 4)

-29 KEY V32E (MARKS = 3)

F 06 37 : : TIG TPI  
 PROF 06 58 HP DELTA VTPI DELTA VTPF  
 PRO

F 06 59 DELTA V'S-LOS

N59 .

DELTA V F/A \_\_\_\_\_

DELTA V R/L \_\_\_\_\_

DELTA V D/U \_\_\_\_\_

PRO

F 16 45 MKS TFI -00001

RESET ET

\*410+3 TPI SEARCH ROUTINE  
 \*307+043.00 DELTA T TRANSFER

-26

\*310+026.00 TFI TPI

\*303R \_\_\_\_\_ THETA AT TPI

\*267R \_\_\_\_\_ DELTA VTPI

\*371R \_\_\_\_\_ DELTA VTPI + DELTA VTPF

KEY V32E (MARKS = 10)

F 06 37 : : TIG TPI

PRO

F 06 58 HP DELTA VTPI DELTA VTPF

PRO

F 06 59 DELTA V\*S-LOS

N59

DELTA V F/A \_\_\_\_\_

DELTA V R/L \_\_\_\_\_

DELTA V D/U \_\_\_\_\_

PRO

F 16 45 MKS TFI -00001

TEMP/PRESS MON SEL - FUEL MAN

A + B PRESS 181 +-3 PSIG

TEMP/PRESS MON SEL - PRPLNT

A + B TEMP IND 40DEG TO 100DEG F

A + B PRESS IND 181 +-3 PSIA

TEMP/PRESS MON SEL - HEL

A + B PRESS IND \_\_\_\_\_ PSIA

A QUANTITY IND \_\_\_\_\_ PERCENT REMAINING

B QUANTITY IND \_\_\_\_\_ PERCENT REMAINING

TEMP MON SEL RCS QUAD 1, 4, 2, 3 (120 - 190 DEG F)

SUIT/CABIN PRESS - NOR

GLYCOL TEMP/PRESS - NOR

O2/H2O QUANTITY - NOMINAL

RESET ET

-13

\*310+013.00 TFI TPI

\*303R \_\_\_\_\_ THETA AT TPI

-12

PRO (FINAL COMP)

F 06 37 : : TIG TPI



F 50 16 DOWNLINK COMPLETE  
PRO (UPDATE COMPLETE)

\*400+3 AGS ALIGN  
\*400R (00000 COMPLETE)

-9\* \*277R COPY TGT LOS ANGLE

\*310+008.00

\*303R \_\_\_\_\_ THETA AT TPI  
\*410+4 DIRECT TRANSFER AT ET=-8  
\*404+0 405+0 406+0  
(MONITOR 470 471 472)

\*373R \_\_\_\_\_ TIG TPI

\*267R \_\_\_\_\_ DELTA V TPI

\*371R \_\_\_\_\_ DELTA V TPI + DELTA V TPP

-6 KEY 41E (RCS THRUSTING)

F 50 18 FDAI ANGLES (REQUEST MNVR)  
ENTR (BYPASS MNVR)

16 85 DELTA V\*S-BODY

KEY V06N86E DELTA V\*S-LV

F 06 86 DELTA V\*S-LV

AGS(N86)	CHART
----------	-------

DELTA VX(LV)	_____	_____
--------------	-------	-------

DELTA VY(LV)	_____	_____
--------------	-------	-------

DELTA VZ(LV)	_____	_____
--------------	-------	-------

KEY RELEASE

-5\* \*277R COPY TGT LOS ANGLE

COPY R, RDOT FROM TAPE METER FOR BU'S

\*400+2 Z-AXIS STEERING  
\*410+5 EXTERNAL DELTA V  
\*450 451 452 LOAD  
\*407+0 ROTATE EXTERNAL DELTA V RF

COMPUTE TPI BU

COPY CSM TPI SOLUTION AND SYNC COUNTDOWN WITH CSM

4-23

16 85 DELTA V'S-BODY

\*500R \_\_\_\_\_ DELTA VX(NOM -11.5)

\*502R \_\_\_\_\_ DELTA VZ(NOM 22.9)

-35 DSKY BLANKS

KEY V77E (ATT HOLD)  
PGNS MODE CONT-ATT HOLD  
AGS MODE CONT-ATT HOLD  
\*407+1 FREEZE EXTERNAL DELTA V

F 16 85 DELTA V'S-BODY

(125:02:46) BURN +2

VERIFY AGS RESIDUALS

\*500R 501R 502R  
DELTA VX \_\_\_\_\_ DELTA VY \_\_\_\_\_ DELTA VZ \_\_\_\_\_

VERIFY PGNS RESIDUALS

N85

DELTA VX(LM) \_\_\_\_\_

DELTA VY(LM) \_\_\_\_\_

DELTA VZ(LM) \_\_\_\_\_

PRO

F 37 BB

\*\*\*\*\*  
\* MCC1 \*  
\*\*\*\*\*

KEY V76E (PGNS PULSE)  
PGNS MODE CONT-AUTO

KEY V93E (BEFORE 1ST MARK REINIT' W MATRIX)

+2 KEY 35E (TPM PRE-THRUST)  
F 16 45 MKS TFI -00001

(LOS OF MSFN)  
 S-BAND - SLEW TO P = +90, Y = 0  
 S-BAND - FWD (OR AFT)  
 S-BAND - DN VOICE BU  
 BIOMED OFF

\*404+0 405+0 406+0  
 (MONITOR 470 471 472)

+9\* \*277R COPY TGT LOS ANGLE

+12 PRO (COMPUTE MCC FOR TPI+15)  
 F 06 81 DELTA V\*S-LV  
 PRO  
 F 06 59 DELTA V\*S-LOS  
 PRO  
 F 16 45 MKS TFI MGA

+13\* \*277R COPY TGT LOS ANGLE  
 COPY R, RDOT FROM TAPE METER FOR BU\*S  
 COMPUTE MCC1 BU  
 PRO  
 F 37 BB

+14 KEY 41E (RCS THRUSTING)  
 F 50 18 FDAO ANGLES (REQUEST MNVR)  
 ENTR (BYPASS MNVR)  
 16 85 DELTA V\*S-BODY

+14:25 DSKY BLANKS  
 -  
 KEY V77E (ATT HOLD)  
 PGNS MODE CONT-ATT HOLD  
 AGS MODE CONT-ATT HOLD  
 \*407+1 FREEZE EXTERNAL DELTA V

+15 F 16 85 NULL DELTA V\*S  
 PRO  
 F 37 BB

\*\*\*\*  
 \* MCC2 \*  
 \*\*\*\*

KEY V76E (PGNS PULSE)

PGNS MODE CONT-AUTO

KEY V93E (BEFORE 1ST MARK REINIT W MATRIX)

+16 KEY 35E (TPM PRE-THRUST)  
F 16 45 MKS TFI -00001\*404+0 405+0 406+0  
(MONITOR 470 471 472)

+24\* \*277R COPY TGT LOS ANGLE

+27 PRO (COMPUTE MCC FOR TPI+30)  
F 06 81 DELTA V\*S-LV  
PRO  
F 06 59 DELTA V\*S-LOS  
PRO  
F 16 45 MKS TFI MGA

+28\* \*277R COPY TGT LOS ANGLE

COPY R, RDOT FROM TAPE METER FOR BU'S

COMPUTE MCC2 BU.

PRO

F 37 BB

KEY 41E (RCS THRUSTING)  
F 50 18 FDAI ANGLES (REQUEST MNVR)  
ENTR (BYPASS MNVR)  
16 85 DELTA V\*S-BODY

+29:25 DSKY BLANKS

KEY V77E (ATT HOLD)  
PGNS MODE CONT-ATT HOLD  
AGS MODE CONT-ATT HOLD  
\*407+1 FREEZE EXTERNAL DELTA V

+30 F 16 85 NULL DELTA V'S

PRO  
F 37 BB

KEY 00E (POO)

V63E (RR SELF TEST)  
F 04 12 00004 00001  
PRO

F 16 72 TRUN SHFT  
PRO  
F 16 78 R RDOT  
KEY V34E (TERMINATE)

KEY V37E47E (AVE G)  
F 16 83 DELTA V'S

V63E (RR SELF TEST)  
F 04 12 00004 00001  
PRO

F 16 72 TRUN SHFT  
PRO  
F 16 78 R RDOT  
VERIFY TAPE METER WITH DSKY

\*\*\*\*\*  
\* BRAKING GATE \*  
\* 30 FPS - 6000 FT \*  
\* 20 FPS - 3000 FT \*  
\* 10 FPS - 1500 FT \*  
\* 5 FPS - 600 FT \*  
\*\*\*\*\*

RENDEZVOUS

STATION KEEPING

5.0 ABORT PROCEDURE

This is a specific AGS nominal abort procedure due to a PGNS failure between insertion and station keeping.

5.1 Nominal AGS Procedures

## ASSUMPTIONS

\*\*\*\*\*  
 \* AFTER INSERTION \*  
 \*\*\*\*\*

- 1 INSERTION COMPLETE (RFSIDUALS NULLED)
- 2 PGNS FAILURE BEFORE INSERTION
- 3 RADAR NOT LOCKED ON

GUID CONT-AGS  
 ATTITUDE MON-AGS(CDR)  
 RATE ERR MON-CMPTR(LMP)  
 ATTITUDE MON-AGS(LMP)  
 SHIFT/TRUN-+5 DEG  
 X-POINTER-HI MULT  
 RADAR TEST SW-OFF  
 ENG GMBL-OFF  
 ENG ARM-OFF  
 X-TRANSL-2 JETS  
 DEAD BAND-MAX  
 PGNS MODE CONT-OFF  
 THROT/JET-JET  
 CB IMU OPR-ON  
 R/R MODE-AUTO TRACK  
 TTCA/TRANSL-ENABLE(CDR)

\*\*\*\*\*  
 \* BAL CPL-ON \*  
 \* MODE SEL-AGS \*  
 \* RNG/ALT MON-RNG/RNGRT \*  
 \* RATE ERR MON-RNDZ RDR(CDR) \*  
 \* ATT CONT(3)-PULSE \*  
 \* AGS MODE CONT-AUTO \*  
 \* ACA/4JET-DISABLE(CDR) \*  
 \* TTCA/TRANSL-DISABLE(LMP) \*  
 \* ACA/4JET-DISABLE(LMP) \*  
 \*\*\*\*\*

\*507+0 Z-AXIS BORESIGHT  
 \*400+2 Z-AXIS STEERING  
 PULSE TO NULL AGS ERRORS  
 POINT Z-AXIS IN DIRECTION OF CSM.  
 MODE CONT-ATT HOLD  
 ATT CONT(3)-MODE CONTROL  
 R/R MODE - SLEW

SLEW RATE - LO  
 SLEW R/R ANT TO 0.0 ON FDAI  
 VERIFY MAX SIGNAL STRENGTH  
 R/R MODE-AUTO TRACK  
 NO TRACK LIGHT OFF  
 ATT CONT(3)-PULSE

\*\*\*\*\*  
 \* CHECK INSERTION VELOCITY WITH \*  
 \* TAPEMETER \*  
 \* COMPARE TAPEMETER WITH AGS STATE \*  
 \* VECTOR DATA \*  
 \* \*317R RANGE (NOM 264.2) \*  
 \* \*440R RANGE RATE (NOM -425.1) \*  
 \*\*\*\*\*

SUIT/CABIN PRESS - NOR  
 GLYCOL TEMP/PRESS - NOR  
 O2/H2O QUANTITY - NOMINAL

\*\*\*\*\*  
 \* CSI \*  
 \*\*\*\*\*

+3 \*417+1 (RADAR FILTER INITIALIZED)  
 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE  
  
 +5 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE  
 BORESIGHT ON CSM  
 \*415+1  
 \*316+XXXX,X RANGF  
  
 \*410+1 CSI ROUTINE  
 \*373+0206.5 CSI TIME  
 \*275+0302.8 TPI TIME  
 \*605+00777 COTANGENT OF LOS TO CSM  
 \*416+1 (ONE-HALF ORBITAL PERIOD)  
 \*411+0 RCS ENGINES  
 \*623+0 Z-AXIS PARALLEL TO CSM ORBIT  
  
 \*310R TIME TO CSI

-43           RESET ET  
  
 \*267R        VELOCITY TO BE GAINED CSI  
  
 \*402R        DFLTA H  
  
 \*371R        DFLTA V CDH

\*372R DFLTA T CSI/CDH  
-42 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
-40 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
BORESIGHT ON CSM  
\*415+1  
\*316+XXXX,X RANGE  
-37 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
-35 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
BORESIGHT ON CSM  
\*415+1  
\*316+XXXX,X RANGE  
-32 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
\*31UR TIME TO CSI  
RESET ET  
\*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
-30\* COPY ROOT FOR 1ST CSI BU  
BORESIGHT ON CSM  
\*415+1  
\*316+XXXX,X RANGE  
-27 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
\*267R VELOCITY TO BE GAINED CSI  
\*402R DELTA H  
\*371R DFLTA V CDH  
\*372R DELTA T CSI/CDH  
-25 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
BORESIGHT ON CSM  
\*415+1  
\*316+XXXX,X RANGE  
-22 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
\*415+1 (NO BORESIGHT)

\*503-XXXX,X RANGE RATE

-20\* COPY RDOT FOR 2ND CSI BU

BORESIGHT ON CSM

\*415+1

\*316+XXXX,X RANGE

TEMP/PRESS MON SEL - FUEL MAN

A AND B PRESS 181 +/- 3 PSIG

TEMP/PRESS MON SEL - PRPLNT

A AND B TEMP IND 40DEG TO 100DEG F

A AND B PRFSS IND 181 +/- 3 PSIA

TEMP/PRESS MON SEL - HEL

A AND B PRESS IND \_\_\_\_\_ PSIA

A QUANTITY IND \_\_\_\_\_ PERCENT RFMAINING

B QUANTITY IND \_\_\_\_\_ PERCENT REMAINING

TEMP MON SEL RCS QUAD 1, 4, 2, 3 (120 - 190 DFG F)

-17

\*415+1 (NO BORESIGHT)

\*503-XXXX,X RANGE RATE

-15

\*415+1 (NO BORESIGHT)

\*503-XXXX,X RANGE RATE

BORESIGHT ON CSM

\*415+1

\*316+XXXX,X RANGE

(LOS OF MSFN)

S-BAND - SLEW TO P = +90, Y = 0

S-BAND - FWD (OR AFT)

S-BAND - DN VOICE BU

BIOMED OFF

-12

\*415+1 (NO BORESIGHT)

\*503-XXXX,X RANGE RATE

\*267R \_\_\_\_\_ VFLOCITY TO BE GAINED CSI

\*402R \_\_\_\_\_ DELTA H

\*372R \_\_\_\_\_ TIME CSI/CDH

\*371R \_\_\_\_\_ PREDICTED VEL TO BE GAINED CDH

\*263R \_\_\_\_\_ OUT-OF-PLANE

\*415+1 (NO BORESIGHT)

\*503-XXXX,X RANGE RATE

-10\*

COPY RANGE AND RDOT FOR 3RD CSI BU

BORESIGHT ON CSM

\*415+1

\*316+XXXX.X RANGE

COMPUTE CSI BU

-7 \*415+1 NO BORESIGHT  
\*503-XXXX.X RANGE RATE

(TRANSMIT CSI DATA TO CSM)  
VHF A XMTR - VOICE  
VHF B XMTR - DATA  
VHF B RCVR - OFF  
TELEMETRY PCM - LO

-5 \*415+1 NO BORESIGHT  
\*503-XXXX.X RANGE RATE  
BORE SIGHT ON CSM  
\*415+1  
\*316+XXXX.X RANGE

\*277R \_\_\_\_\_ THETA  
SET ORDEAL

\*317R \_\_\_\_\_ RANGE

\*440R \_\_\_\_\_ RANGE RATE

\*507+1 Z-AXIS DELTA V  
\*400+0 ATT HOLD  
\*410+5 EXTERNAL DELTA V  
(LOAD SOLUTION TO BE EXECUTED)  
\*450 451 452  
\*407+0 ROTATE EXTERNAL DELTA V RF

NULL FRROR SIGNALS WITH AGS PULSE

TRANSMIT TGT DELTA V'S TO CSM AND SYNC COUNTDOWN

\*502R \_\_\_\_\_ DFLTA VZ (NOM 50.1)

-:20 DEADBAND-MIN  
(DEFLECT RHC OUT OF DETENT)  
AGS MODE CONT-ATT HOLD  
ATT CONT(3)-MODE CONT  
\*407+1 FREEZE EXTERNAL DELTA V

SET ET COUNTING UP AT ZERO

(123:26:27) BURN +Z

NULL DELTA V'S

VERIFY AGS RESIDUALS

\*500R 501R 502R  
DELTA VX \_\_\_\_\_ DELTA VY \_\_\_\_\_ DELTA VZ \_\_\_\_\_

\*\*\*\*\*  
\* CDH \*  
\*\*\*\*\*

ATT CONT(3)-PULSE  
AGS MODE CONT-AUTO  
\*507+0 Z-AXIS BORESIGHT  
\*400+2 Z-AXIS STEERING

+5 \*410+2 CDH ROUTINE

(VHF RANGING WITH CSM)  
VHF A XMTR = VOICE/RNG  
VHF B XMTR = OFF  
VHF B RCVR = ON  
TELEMFTRY PCM = HI

\*411+0 THRUST ALONG Z-BODY AXIS

\*373R \_\_\_\_\_ (NOM 264.4)  
\*310R TIME TO CDH

-52 RESET ET

\*423R \_\_\_\_\_ DESIRED ALT RATE AT CDH

\*402R \_\_\_\_\_ DELTA H

\*450R \_\_\_\_\_ DELTA VX

\*452R \_\_\_\_\_ DELTA VZ  
\*623+0 Z-AXIS PARALLEL TO CSM PLANE

-50 \*417+1 (RADAR FILTER INITIALIZED)  
\*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
BORESIGHT ON CSM  
\*415+1  
\*316+XXXX,X RANGE

-47 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE

-45 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE  
BORESIGHT ON CSM  
\*415+1  
\*316+XXXX,X RANGE

-42 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE

-40 \*415+1 (NO BORESIGHT)  
\*503-XXXX,X RANGE RATE

BORESIGHT ON CSM  
 \*415+1  
 \*316+XXXX,X RANGF

\*\*\*\*\*  
 \* PLANE CHANGE \*  
 \*\*\*\*\*

\*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE

-36\* COPY RDOT FOR 1ST CDH BU

-35 BORESIGHT ON CSM  
 \*415+1  
 \*316+XXXX,X RANGF

\*507+1 Z-AXIS DELTA V  
 \*400+0 ATT HOLD  
 \*410+5 EXTERNAL DELTA V  
 (COPY YDOT FROM CSM AND LOAD NEGATIVE IN 451)  
 \*450+0 451 452+0  
 \*407+0 ROTATE EXTERNAL DELTA V RF

NULL ERROR SIGNALS WITH AGS PULSE

-2u DEADBAND-MIN  
 (DEFLECT RHC OUT OF DETENT)  
 AGS MODE CONT-ATI HOLD  
 ATT CONT(3)-MODF CONT  
 \*407+1 FREEZE EXTERNAL DELTA V

-30 NULL DELTA V'S

ATT CONT(3)-PULSE  
 AGS MODE CONT-AUTO  
 \*507+0 Z-AXIS BORESIGHT  
 \*400+2 Z-AXIS STFERING

\*410+2 CDH ROUTINE

\*373R\_\_\_\_\_ (NUM 264.4)  
 \*310R TIME TO CDH

\*423R\_\_\_\_\_ DESIRED ALT RATE AT CDH

\*402R\_\_\_\_\_ DELTA H

\*450R\_\_\_\_\_ DELTA VX

\*452R\_\_\_\_\_ DELTA VZ

- 27
  - \*417+1 (RADAR FILTER INITIALIZFD)
  - \*415+1 (NO BORESIGHT)
  - \*503-XXXX.X RANGE RATE
- 25
  - \*415+1 (NO BORESIGHT)
  - \*503-XXXX.X RANGE RATE
  - BORESIGHT ON CSM
  - \*415+1
  - \*316+XXXX.X RANGE

(AOS OF MSFN)  
 S=BAND - SLEW TO P = \_\_\_, Y = \_\_\_,  
 ACQUIRE HGA LOCK ON  
 VOICE - BU/VOICE -- VOICE  
 VERIFY VOICE, TLM  
 BIOMED - LEFT (OR RIGHT)

  - \*415+1 (NO BORESIGHT)
  - \*503-XXXX.X RANGE RATE
- 23\*
  - COPY RDOT FOR 2ND CDH BU
  - TEMP/PRESS MON SEL - FUEL MAN  
 A AND B PRESS 181 +-3 PSIG
  - TEMP/PRESS MON SEL - PRPLNT  
 A AND B TEMP IND 40DEG TO 100DEG F
  - A AND B PRFSS IND 181 +-3 PSIA
  - TEMP/PRESS MON SEL - HEL
  - A AND B PRESS IND \_\_\_\_\_ PSIA
  - A QUANTITY IND \_\_\_\_\_ PERCENT REMAINING
  - B QUANTITY IND \_\_\_\_\_ PERCENT REMAINING
  - TEMP MON SEL RCS QUAD 1, 4, 2, 3 (120 - 190 DFG F)
  - SUIT/CABIN PRESS = NOR
  - GLYCOL TEMP/PRESS = NOR
  - O2/H2O QUANTITY - NOMINAL
- 20
  - \*415+1 (NO BORESIGHT)
  - \*503-XXXX.X RANGE RATE
  - BORESIGHT ON CSM
  - \*415+1
  - \*316+XXXX.X RANGE
- 17
  - \*415+1 (NO BORESIGHT)
  - \*503-XXXX.X RANGE RATE
- 15
  - \*415+1 (NO BORESIGHT)
  - \*503-XXXX.X RANGE RATE
  - BORESIGHT ON CSM
  - \*415+1
  - \*316+XXXX.X RANGE
- 12
  - \*415+1 (NO BORESIGHT)
  - \*503-XXXX.X RANGE RATE

\*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE

-10\* COPY RDOT FOR 3RD CDH BU

BORESIGHT ON CSM  
 \*415+1  
 \*316+XXXX,X RANGE

COMPUTE CDH BU

-7 \*415+1 NO BORESIGHT  
 \*503-XXXX,X RANGE RATE

-5 \*415+1 NO BORESIGHT  
 \*503-XXXX,X RANGE RATE  
 BORESIGHT ON CSM  
 \*415+1  
 \*316+XXXX,X RANGE

\*277R \_\_\_\_\_ THETA  
 SET ORDEAL

\*317R \_\_\_\_\_ RANGE

\*440R \_\_\_\_\_ RANGE RATE

\*507+1 Z-AXIS DELTA V  
 \*400+0 ATT HOLD  
 \*410+5 EXTERNAL DELTA V  
 (LOAD SOLUTION TO BE EXECUTED)  
 \*450 451 452  
 \*407+0 ROTATE EXTERNAL DELTA V RF

NULL ERROR SIGNALS WITH AGS PULSE

TRANSMIT TGT DELTA V'S TO CSM AND SYNC COUNTDOWN

\*500R \_\_\_\_\_ DELTA VX (NOM -0.3)

\*502R \_\_\_\_\_ DELTA VZ (NOM 6.0)

-:20 DEADBAND-MIN  
 (DEFLECT RHC OUT OF DETFNT)  
 AGS MODE CONT-ATT HOLD  
 ATT CONT(3)-MODE CONT  
 \*407+1 FREEZE EXTERNAL DELTA V

SET ET COUNTING UP AT ZERO

(124:24:25) NULL DELTA V'S

VERIFY AGS RESIDUALS

\*500R 501R 502R

DELTA VX \_\_\_\_\_ DELTA VY \_\_\_\_\_ DELTA VZ \_\_\_\_\_

\*\*\*\*\*  
 \* TPI \*  
 \*\*\*\*\*

ATT CONT(3)-PULSE  
 AGS MODE CONT-AUTO  
 \*507+0 Z-AXIS BORESIGHT  
 \*400+2 Z-AXIS STEERING

\*417+1 (RADAR FILTER INITIALIZED)  
 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE

+3 \*410+3 TPI SEARCH ROUTINE  
 \*307+043.00 DELTA T TRANSFER  
 \*310+034.00 TFI TPI

\*303R \_\_\_\_\_ THFTA AT TPI  
 \*410+4 (WHEN 303 = 26.60 DEG)

\*310R \_\_\_\_\_ TFI TPI

RESET ET

\*267R \_\_\_\_\_ DELTA V TPI

\*371R \_\_\_\_\_ DELTA V TPI + DELTA V TPF

-32 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE

-30 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE  
 BORESIGHT ON CSM  
 \*415+1  
 \*316+XXXX,X RANGE

-27 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE

-25 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE  
 BORESIGHT ON CSM  
 \*415+1  
 \*316+XXXX,X RANGE

-22 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE

\*410+3 TPI SEARCH ROUTINE  
 \*307+043.00 DELTA T TRANSFER  
 \*310+021.00 TFI TPI

\*303R THETA AT TPI  
 \*410+4 (WHEN 303 = 26.60 DEG)

\*310R TFI TPI

RESET ET

\*267R DELTA V TPI

\*371R DELTA V TPI + DELTA V TPF

-20 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE  
 BORESIGHT ON CSM  
 \*415+1  
 \*316+XXXX,X RANGE

-17 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE

TEMP/PRESS MON SEL - FUEL MAN  
 A AND B PRESS 181 +-3 PSIG

TEMP/PRESS MON SEL - PRPLNT  
 A AND B TEMP IND 40DEG TO 100DEG F  
 A AND B PRESS IND 181 +-3 PSIA

TEMP/PRESS MON SEL - HEL

A AND B PRESS IND PSIA

A QUANTITY IND PERCENT REMAINING

B QUANTITY IND PERCENT REMAINING

TEMP MON SEL RCS QUAD 1, 4, 2, 3 (120 - 190 DEG F)

SUIT/CABIN PRESS - NOR

GLYCOL TEMP/PRESS - NOR

O2/H2O QUANTITY - NOMINAL

-15 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE  
 BORESIGHT ON CSM  
 \*415+1  
 \*316+XXXX,X RANGE

-12 \*415+1 (NO BORESIGHT)  
 \*503-XXXX,X RANGE RATE

\*410+3 TPI SEARCH ROUTINE  
 \*307+043.00 DELTA T TRANSFER

-11 \*310+011.00 TFI TPI

\*303R THETA AT TPI  
 \*410+4 (WHEN 303 = 26.60 DEG)

\*277R THETA  
 SET ORDEAL

\*310R TFI TPI

RESET ET

\*267R DELTA V TPI

\*371R DELTA V TPI + DELTA V TPF

TRANSMIT TIG OF TPI TO CSM

-9\* \*277R COPY TGT LOS ANGLE

-7 \*415+1 NO BORESIGHT  
\*503-XXXX,X RANGE RATE

\*404+0 405+0 406+0  
(MONITOR 470 471 472)

\*415+1 NO BORESIGHT  
\*503-XXXX,X RANGE RATE

-5\* \*277R COPY TGT LOS ANGLE

COPY R, RDOT FROM TAPE METER FOR BU'S

BORESIGHT ON CSM

\*415+1  
\*316+XXXX,X RANGE

\*373R TIG TPI

\*267R DELTA V TPI

COMPUTE TPI BU

\*371R DELTA V TPI + DELTA V TPF

\*507+1 Z-AXIS DELTA V

\*400+0 ATT HOLD

\*410+5 EXTERNAL DELTA V  
(LOAD SOLUTION TO BE EXECUTED)

\*450 451 452

\*407+0 ROTATE EXTERNAL DELTA V RF

NULL ERROR SIGNALS WITH AGS PULSE

COPY CSM TPI SOLUTION AND SYNC COUNTDOWN WITH CSM

\*500R DELTA VX (NOM -11.5)

\*502R DELTA VZ (NOM 22.9)

-:20 DEADBAND-MIN  
(DEFLECT RHC OUT OF DETENT)  
AGS MODE CONT-ATT HOLD  
ATT CONT(3)-MODE CONT

\*407+1 FREEZE EXTERNAL DELTA V

SET ET COUNTING UP AT ZERO

(125:02:46) BURN +2

VERIFY AGS RESIDUALS

\*500R 501R 502R  
DELTAVX DELTAVY DELTAVZ

\*\*\*\*\*  
\* MCC1 \*  
\*\*\*\*\*

ATT CONT(3)-PULSE  
AGS MODE CONT-AUTO  
\*507+0 Z-AXIS BORESIGHT  
\*400+2 Z-AXIS STEERING

(LOS OF MSFN)  
S-BAND - SLEW TO P = +90, Y = 0  
S-BAND - FWD (OR AFT)  
S-BAND - DN VOICE BU  
BIOMED OFF

\*404+0 405+0 406+0

+9\* \*277R COPY TGT LOS ANGLE

+13\* \*277R COPY TGT LOS ANGLE

COPY R, RDOT FROM TAPE METER FOR BU'S

COMPUTE MCC1 BU

+14.40 DEADBAND-MIN  
(DEFLECT RHC OUT OF DETENT)  
AGS MODE CONT-ATT HOLD  
ATT CONT(3)-MODF CONT  
\*400+0 ATT HOLD  
\*470R 471R 472R

+15 NULL DELTA V S

\*\*\*\*\*  
\* MCC2 \*  
\*\*\*\*\*

ATT CONT(3)-PULSE  
AGS MODE CONT-AUTO  
\*404+0 405+0 406+0

+24\* \*277R COPY TGT LOS ANGLE

+28\* \*277R COPY TGT LOS ANGLE

COPY R, RDOT FROM TAPE METER FOR BU'S

COMPUTE MCC2 RU

+29,40 DEADBAND-MIN  
(DEFLECT RHC OUT OF DETENT)  
AGS MODE CONT-ATT HOLD  
ATT CONT(3)-MODE CONT  
\*470R 471R 472R

+30 NULL DELTA V S

ATT CONT(3)-PULSE  
VERIFY TAPE METER WITH AGS

LOS CONTROL

\*\*\*\*\*  
\* BRAKING GATE \*  
\* 30 FPS - 6000 FT \*  
\* 20 FPS - 3000 FT \*  
\* 10 FPS - 1500 FT \*  
\* 5 FPS - 600 FT \*  
\*\*\*\*\*

RENDEZVOUS

STATION KEEPING

6.0 ONBOARD RENDEZVOUS CHARTS

Charts 6.1 thru 6.6 are the nominal onboard rendezvous charts for Mission G

R1	F1	R2	F2	R3	F3	R3	F4
-240.0	247.3	-140.0	254.7	-70.0	72.4	120.0	15.2
-241.0	248.4	-141.0	256.6	-71.0	73.4	121.0	15.5
-242.0	249.4	-142.0	258.4	-72.0	74.4	122.0	15.7
-243.0	250.5	-143.0	260.2	-73.0	75.5	123.0	15.9
-244.0	251.5	-144.0	262.1	-74.0	76.5	124.0	16.1
-245.0	252.5	-145.0	263.9	-75.0	77.5	125.0	16.3
-246.0	253.6	-146.0	265.7	-76.0	78.6	126.0	16.5
-247.0	254.6	-147.0	267.6	-77.0	79.6	127.0	16.7
-248.0	255.7	-148.0	269.4	-78.0	80.6	128.0	16.9
-249.0	256.7	-149.0	271.3	-79.0	81.7	129.0	17.1
-250.0	257.8	-150.0	273.1	-80.0	82.7	130.0	17.3
-251.0	258.8	-151.0	274.9	-81.0	83.7	131.0	17.5
-252.0	259.9	-152.0	276.8	-82.0	84.8	132.0	17.7
-253.0	260.9	-153.0	278.6	-83.0	85.8	133.0	17.9
-254.0	262.0	-154.0	280.4	-84.0	86.9	134.0	18.1
-255.0	263.0	-155.0	282.3	-85.0	87.9	135.0	18.4
-256.0	264.1	-156.0	284.1	-86.0	88.9	136.0	18.6
-257.0	265.1	-157.0	286.0	-87.0	90.0	137.0	18.8
-258.0	266.2	-158.0	287.8	-88.0	91.0	138.0	19.0
-259.0	267.2	-159.0	289.7	-89.0	92.0	139.0	19.2
-260.0	268.3	-160.0	291.5	-90.0	93.1	140.0	19.4
-261.0	269.3	-161.0	293.4	-91.0	94.1	141.0	19.6
-262.0	270.4	-162.0	295.2	-92.0	95.2	142.0	19.8
-263.0	271.4	-163.0	297.0	-93.0	96.2	143.0	20.0
-264.0	272.5	-164.0	298.9	-94.0	97.2	144.0	20.2
-265.0	273.5	-165.0	300.7	-95.0	98.3	145.0	20.5
-266.0	274.6	-166.0	302.6	-96.0	99.3	146.0	20.7
-267.0	275.7	-167.0	304.4	-97.0	100.4	147.0	20.9
-268.0	276.7	-168.0	306.3	-98.0	101.4	148.0	21.1
-269.0	277.8	-169.0	308.1	-99.0	102.4	149.0	21.3
-270.0	278.8	-170.0	310.0	-100.0	103.5	150.0	21.5
-271.0	279.9	-171.0	311.9	-101.0	104.5	151.0	21.7
-272.0	281.0	-172.0	313.7	-102.0	105.6	152.0	21.9
-273.0	282.0	-173.0	315.6	-103.0	106.6	153.0	22.1
-274.0	283.1	-174.0	317.4	-104.0	107.7	154.0	22.4
-275.0	284.2	-175.0	319.3	-105.0	108.7	155.0	22.6
-276.0	285.2	-176.0	321.1	-106.0	109.7	156.0	22.8
-277.0	286.3	-177.0	323.0	-107.0	110.8	157.0	23.0
-278.0	287.4	-178.0	324.9	-108.0	111.8	158.0	23.2
-279.0	288.4	-179.0	326.7	-109.0	112.9	159.0	23.4
-280.0	289.5	-180.0	328.6	-110.0	113.9	160.0	23.6
-281.0	290.6	-181.0	330.4	-111.0	115.0	161.0	23.8

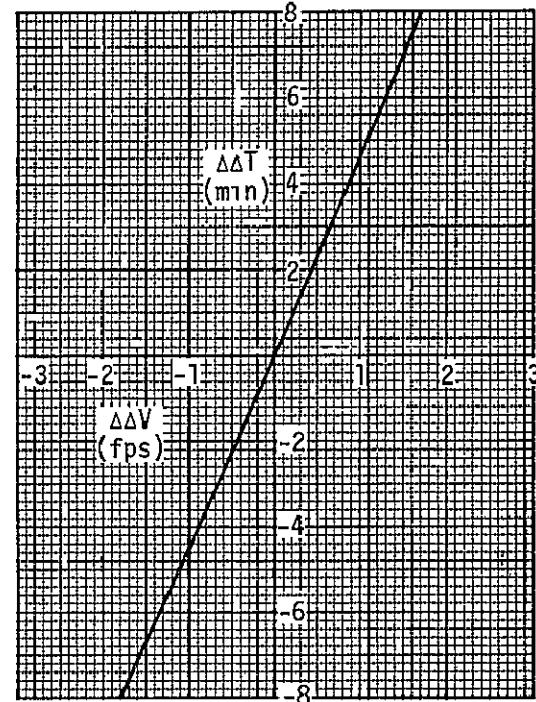
 CSI BACKUP TABLE  
 MISSION G

TIME (Min)	NOMINAL
-30 R1	(-283.3)
-20 R2	(-173.9)
-10 R3	(- 94.0)
-10 R3	( 154.1)
F1	( 293.0)
+F3	( 97.2)
	( 390.2)
-F2	(-317.3)
	( 72.9)
-F4	(- 22.4)
	( 50.5)
+ VCSI	( 0.0)
VCSI	( 50.5)

PREPARED by FPrB/OPS

MISSION APOLLO 11, MAY 16, 1969

R1	F1	R2	F2	R3	F3	R3	F4
-282.0	291.6	-182.0	332.3	-112.0	116.0	162.0	24.0
-283.0	292.7	-183.0	334.2	-113.0	117.0	163.0	24.2
-284.0	293.8	-184.0	336.0	-114.0	118.1	164.0	24.5
-285.0	294.8	-185.0	337.9	-115.0	119.1	165.0	24.7
-286.0	295.9	-186.0	339.8	-116.0	120.2	166.0	24.9
-287.0	297.0	-187.0	341.6	-117.0	121.2	167.0	25.1
-288.0	298.1	-188.0	343.5	-118.0	122.3	168.0	25.3
-289.0	299.2	-189.0	345.4	-119.0	123.3	169.0	25.5
-290.0	300.2	-190.0	347.2	-120.0	124.4	170.0	25.7
-291.0	301.3	-191.0	349.1	-121.0	125.4	171.0	25.9
-292.0	302.4	-192.0	351.0	-122.0	126.5	172.0	26.1
-293.0	303.5	-193.0	352.9	-123.0	127.5	173.0	26.3
-294.0	304.6	-194.0	354.7	-124.0	128.6	174.0	26.5
-295.0	305.6	-195.0	356.6	-125.0	129.6	175.0	26.7
-296.0	306.7	-196.0	358.5	-126.0	130.7	176.0	26.9
-297.0	307.8	-197.0	360.4	-127.0	131.7	177.0	27.1
-298.0	308.9	-198.0	362.2	-128.0	132.8	178.0	27.3
-299.0	310.0	-199.0	364.1	-129.0	133.8	179.0	27.5
-300.0	311.1	-200.0	366.0	-130.0	134.9	180.0	27.7
-301.0	312.2	-201.0	367.9	-131.0	135.9	181.0	27.9
-302.0	313.2	-202.0	369.8	-132.0	137.0	182.0	28.1
-303.0	314.3	-203.0	371.6	-133.0	138.0	183.0	28.3
-304.0	315.4	-204.0	373.5	-134.0	139.1	184.0	28.5
-305.0	316.5	-205.0	375.4	-135.0	140.1	185.0	28.7
-306.0	317.6	-206.0	377.3	-136.0	141.2	186.0	28.9
-307.0	318.7	-207.0	379.2	-137.0	142.2	187.0	29.1
-308.0	319.8	-208.0	381.1	-138.0	143.3	188.0	29.3
-309.0	320.9	-209.0	383.0	-139.0	144.3	189.0	29.5
-310.0	322.0	-210.0	384.9	-140.0	145.4	190.0	29.7
-311.0	323.1	-211.0	386.8	-141.0	146.4	191.0	29.9
-312.0	324.2	-212.0	388.7	-142.0	147.5	192.0	30.1
-313.0	325.3	-213.0	390.5	-143.0	148.5	193.0	30.3
-314.0	326.4	-214.0	392.4	-144.0	149.6	194.0	30.5
-315.0	327.5	-215.0	394.3	-145.0	150.7	195.0	30.7
-316.0	328.6	-216.0	396.2	-146.0	151.7	196.0	30.9
-317.0	329.7	-217.0	398.1	-147.0	152.8	197.0	31.0
-318.0	330.8	-218.0	400.0	-148.0	153.8	198.0	31.2
-319.0	331.9	-219.0	401.9	-149.0	154.9	199.0	31.4
-320.0	333.0	-220.0	403.8	-150.0	155.9	200.0	31.6
-321.0	334.1	-221.0	405.7	-151.0	157.0	201.0	31.8
-322.0	335.3	-222.0	407.7	-152.0	158.1	202.0	32.0
-323.0	336.4	-223.0	409.6	-153.0	159.1	203.0	32.2



TIG TPI	—	—
- TIG TPI	—	—
ΔT	—	—
- NOM ΔT	—	35:21
ΔΔT	—	—
ΔVCSI	—	—

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RDOT	X1	Z1	X2	Z2	X3	Z3
-75.	29.7	52.6	30.6	132.0	.9	79.2
-76.	30.1	53.4	31.0	133.9	.9	80.3
-77.	30.5	54.3	31.3	135.9	.9	81.5
-78.	30.8	55.1	31.7	137.9	.9	82.6
-79.	31.2	55.9	32.1	139.9	.9	83.8
-80.	31.6	56.8	32.5	141.8	.9	84.9
-81.	32.0	57.6	32.8	143.8	.9	86.1
-82.	32.4	58.4	33.2	145.8	.9	87.2
-83.	32.7	59.3	33.6	147.8	.8	88.4
-84.	33.1	60.1	33.9	149.8	.8	89.6
-85.	33.5	61.0	34.3	151.8	.8	90.7
-86.	33.9	61.8	34.7	153.8	.8	91.9
-87.	34.2	62.7	35.0	155.9	.8	93.1
-88.	34.6	63.5	35.4	157.9	.8	94.2
-89.	35.0	64.4	35.8	159.9	.8	95.4
-90.	35.4	65.3	36.1	161.9	.8	96.6
-91.	35.7	66.1	36.5	164.0	.8	97.8
-92.	36.1	67.0	36.9	166.0	.8	98.9
-93.	36.5	67.9	37.2	168.1	.7	100.1
-94.	36.9	68.8	37.6	170.0	.7	101.3
-95.	37.2	69.6	38.0	172.2	.7	102.5
-96.	37.6	70.5	38.3	174.2	.7	103.7
-97.	38.0	71.4	38.7	176.3	.7	104.9
-98.	38.3	72.3	39.0	178.4	.7	106.1
-99.	38.7	73.2	39.4	180.5	.6	107.3
-100.	39.1	74.1	39.8	182.6	.6	108.5
-101.	39.5	75.0	40.1	184.7	.6	109.7
-102.	39.8	75.9	40.5	186.7	.6	110.9
-103.	40.2	76.8	40.8	188.9	.6	112.1
-104.	40.6	77.7	41.2	191.0	.6	113.3
-105.	40.9	78.6	41.5	193.1	.5	114.5
-106.	41.3	79.5	41.9	195.2	.5	115.7
-107.	41.7	80.4	42.2	197.3	.5	116.9
-108.	42.0	81.4	42.6	199.4	.5	118.2
-109.	42.4	82.3	42.9	201.6	.5	119.4
-110.	42.8	83.2	43.3	203.7	.4	120.6
-111.	43.1	84.2	43.6	205.9	.4	121.8
-112.	43.5	85.1	44.0	208.0	.4	123.0
-113.	43.9	86.0	44.3	210.2	.4	124.3
-114.	44.2	87.0	44.7	212.3	.4	125.5
-115.	44.6	87.9	45.0	214.5	.3	126.7
-116.	45.0	88.9	45.4	216.7	.3	128.0
-117.	45.3	89.8	45.7	218.8	.3	129.2

CDH BACKUP TABLE  
MISSION G

	TIME (MIN)	NOMINAL
-36	R1	_____ (-122.21)
-23	R2	_____ (-122.68)
-10	R3	_____ (-122.87)
VX	X1	_____ ( 47.2 )
	+X3	_____ ( 0.1 )
		_____ ( 47.3 )
	-X2	_____ ( -47.7 )
	VX	_____ (- 0 4 ) FPS
VZ	Z2	_____ ( 231.3 )
	-Z1	_____ ( -94.8 )
		_____ ( 136.5 )
	-Z3	_____ (-136.5 )
	VZ	_____ ( 0.0 ) FPS

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## CHART 6 2

RDOT	X1	Z1	X2	Z2	X3	Z3
-118.	45.7	90.8	46 1	221 0	.3	130.5
-119.	46 1	91.7	46 4	223.2	.2	131.7
-120.	46.4	92.7	46.7	225.4	.2	132.9
-121.	46 8	93.6	47.1	227.6	.2	134 2
-122.	47.1	94.6	47.4	229.8	.2	135 4
-123.	47.5	95.6	47.8	232.0	.1	136.7
-124.	47.9	96.5	48.1	234.2	.1	137.9
-125.	48.2	97.5	48 4	236.4	.1	139.2
-126.	48.6	98.5	48 8	238.6	.0	140.5
-127.	48.9	99 5	49 1	240.9	.0	141 7
-128.	49 3	100.5	49.5	243.1	-.0	143.0
-129.	49.7	101.5	49.8	245.3	-.0	144 3
-130.	50.0	102.4	50.1	247.6	-.1	145.5
-131.	50.4	103.4	50.5	249.8	-.1	146 8
-132.	50.7	104.4	50.8	252.1	-.1	148.1
-133.	51.1	105.4	51.1	254.4	-.2	149.3
-134.	51.5	106.4	51.4	256 6	-.2	150.6
-135.	51.8	107.4	51.8	258.9	-.2	151 9
-136.	52.2	108.5	52.1	261.2	-.3	153 2
-137.	52.5	109.5	52.4	263.4	-.3	154 5
-138.	52.9	110.5	52 8	265.7	-.3	155.8
-139.	53.2	111.5	53 1	268.0	-.4	157 0
-140.	53.6	112.5	53.4	270.3	-.4	158 3
-141.	53 9	113.6	53 7	272.6	-.4	159.6
-142.	54.3	114.6	54.1	274.9	-.5	160.9
-143.	54.6	115.6	54.4	277.2	-.5	162.2
-144.	55.0	116 6	54.7	279.6	-.5	163.5
-145.	55.3	117 7	55.0	281.9	-.6	164.8
-146.	55.7	118.7	55.3	284.2	-.6	166.1
-147.	56.1	119.8	55.7	286.5	-.7	167.4
-148.	56 4	120.8	56.0	288.9	-.7	168.8
-149.	56.8	121.9	56.3	291.2	-.7	170.1
-150.	57.1	122.9	56.6	293.6	-.8	171.4
-151.	57.5	124 0	56.9	295.9	-.8	172.7
-152.	57.8	125 0	57 3	298.3	-.9	174.0
-153.	58.2	126.1	57.6	300 7	-.9	175.3
-154.	58.5	127.2	57.9	303.0	-.9	176.7
-155.	58.8	128.2	58.2	305.4	-1.0	178.0
-156.	59.2	129.3	58.5	307.8	-1.0	179 3
-157.	59.5	130.4	58 8	310.2	-1.1	180.7
-158.	59.9	131 5	59 1	312.6	-1.1	182 0
-159.	60.2	132.6	59.4	315.0	-1 2	183.3

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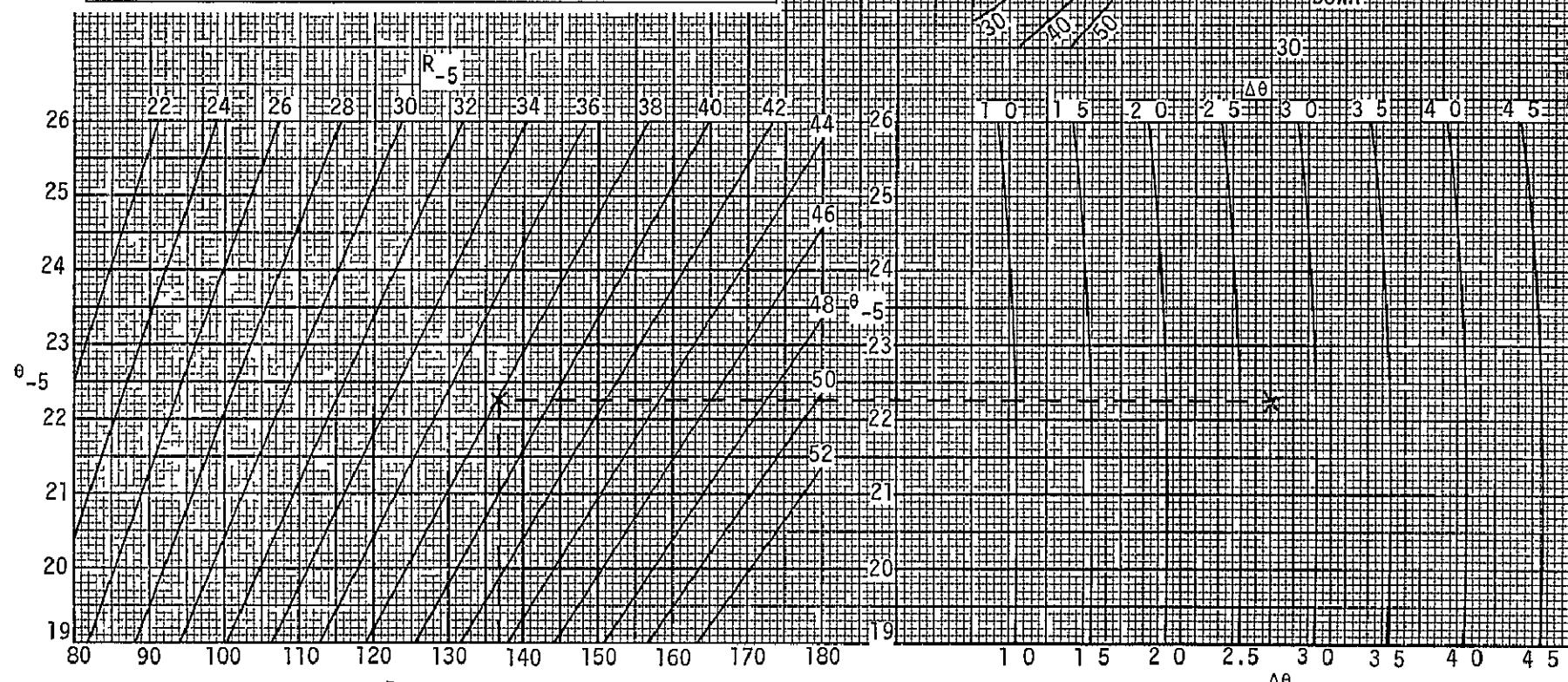
## G MISSION TERMINAL PHASE INITIATION

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CHART 6 3

$\theta_{-5}$  22 24  $R_{-5}$  38.18  $R_{-5}$  112 3  
 $\theta_{-9}$  19 54  $\Delta R$  24 6  
 $\Delta\theta$  2.70

PNGS (N59)	GND	CHARTS	$\Delta T$
$\pm A$ _____	_____	_____	_____
$\pm L$ _____	_____	_____	_____
$\pm U$ _____	_____	_____	_____



MISSION APOLLO 11, MAY 16, 1969

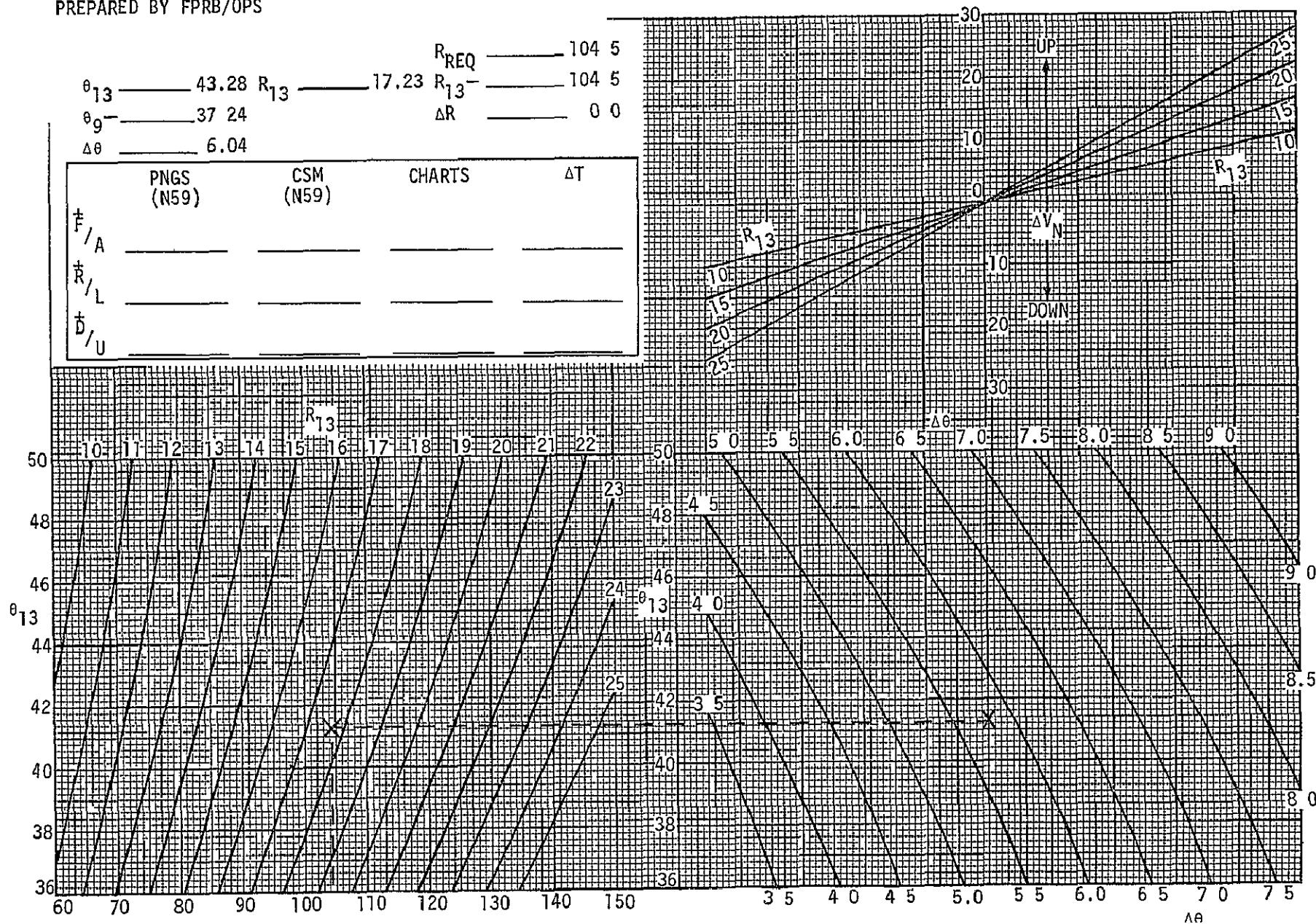
## G MISSION FIRST MIDCOURSE CORRECTION

CHART 6 4

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$\theta_{13}$  — 43.28  $R_{13}$  — 17.23  $R_{13}$  — 104.5  
 $\theta_9$  — 37.24  $\Delta R$  — 0.0  
 $\Delta\theta$  — 6.04

	PNGS (N59)	CSM (N59)	CHARTS	$\Delta T$
$F/A$	—	—	—	—
$R/L$	—	—	—	—
$D/U$	—	—	—	—



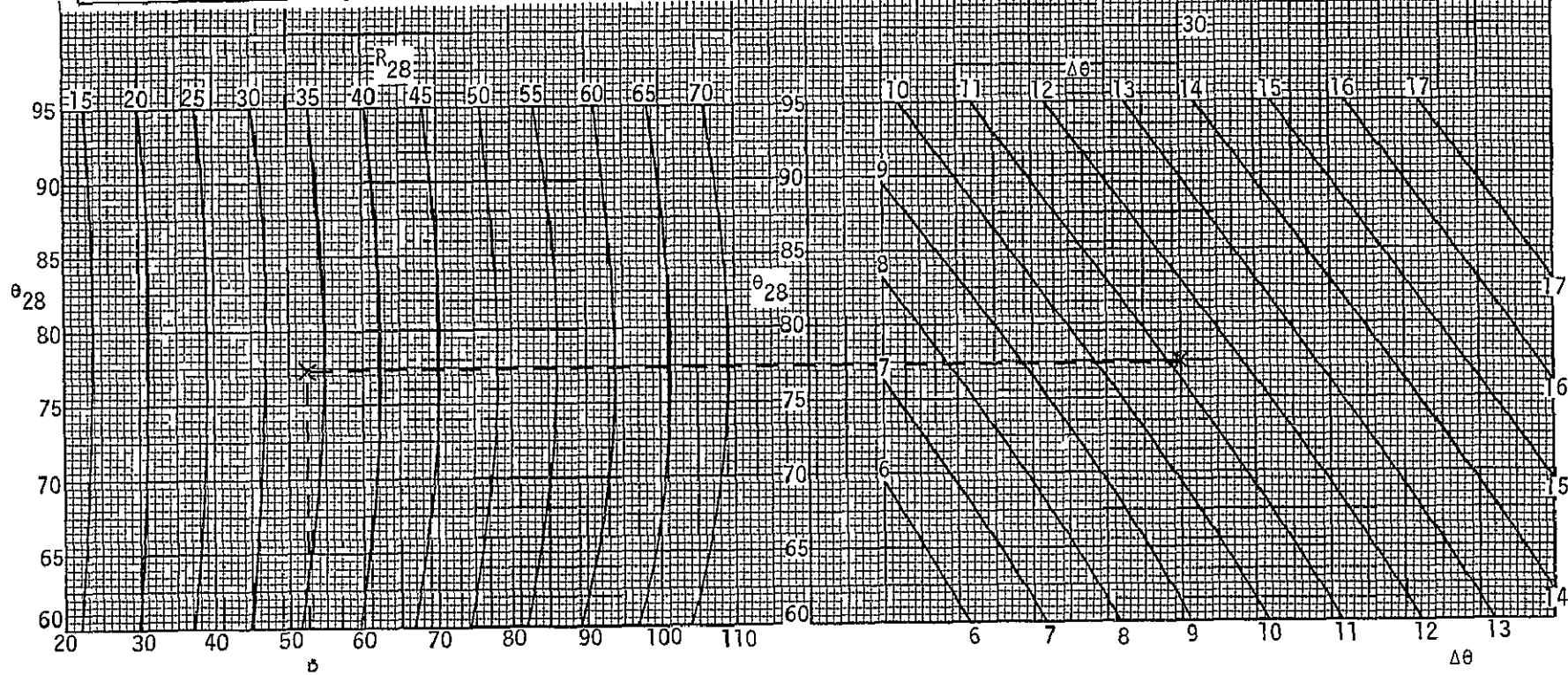
G MISSION SECOND MIDCOURSE CORRECTION

PREPARED BY FPRB/OPS

$\theta_{28}$	77 41	$R_{28}$	34 3	$R_{28}$	52 4
$\theta_{24}$	66 22			$\Delta R$	0.0
$\Delta\theta$	11 19				

CHART 6.5

	PNGS (N59)	CSM (N59)	CHARTS	$\Delta T$
$\dot{\theta}/A$	_____	_____	_____	_____
$\dot{R}/L$	_____	_____	_____	_____
$\dot{d}/U$	_____	_____	_____	_____

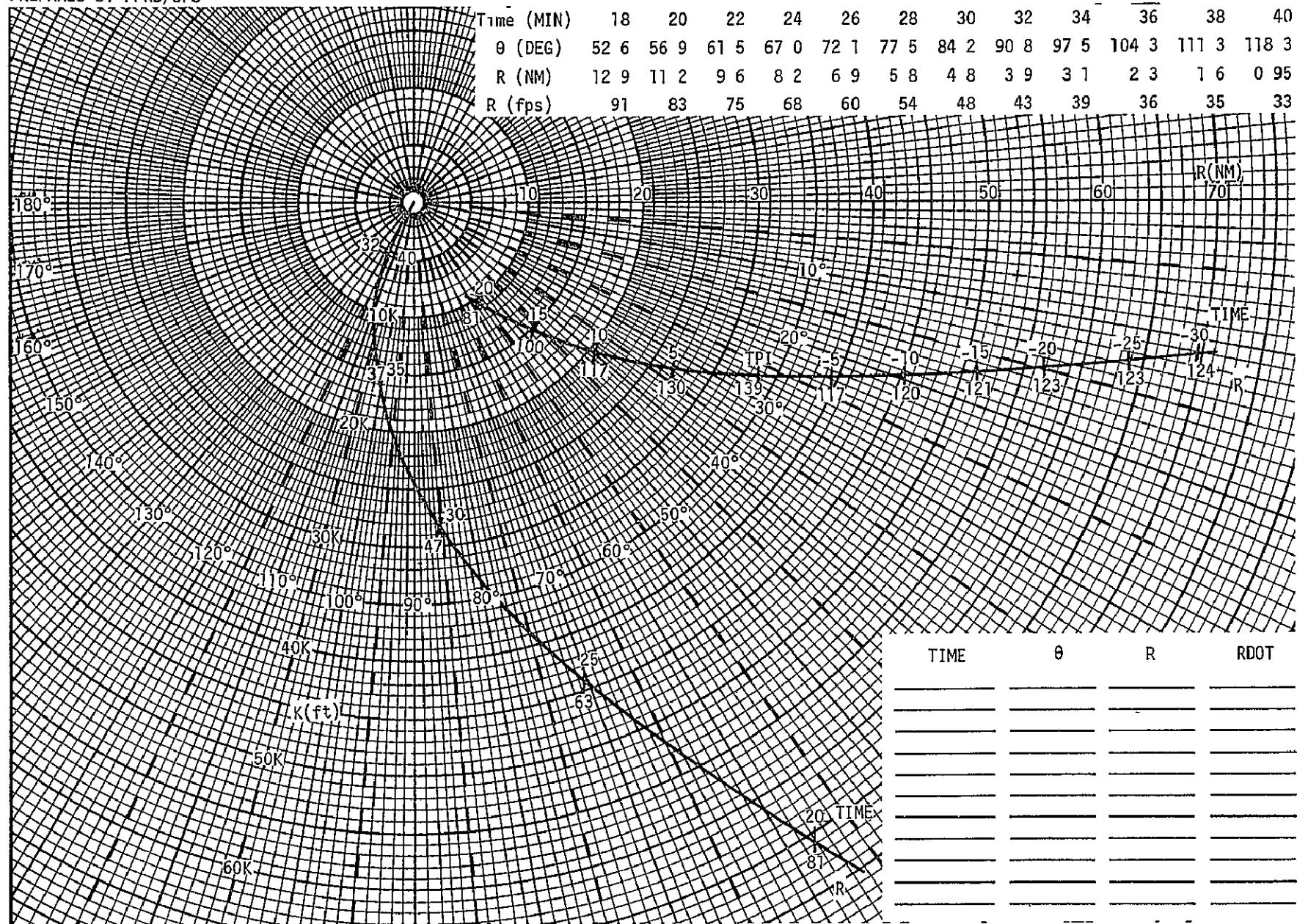


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## G MISSION RELATIVE REFERENCE TRAJECTORY

CHART 6 6

Time (MIN)	18	20	22	24	26	28	30	32	34	36	38	40
$\theta$ (DEG)	52.6	56.9	61.5	67.0	72.1	77.5	84.2	90.8	97.5	104.3	111.3	118.3
R (NM)	12.9	11.2	9.6	8.2	6.9	5.8	4.8	3.9	3.1	2.3	1.6	0.95
R (fps)	91	83	75	68	60	54	48	43	39	36	35	33



MISSION APOLLO 11, MAY 16, 1969

7.0 REFERENCES

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- 7.3 Pixley, P. T.; F Rendezvous Navigation Mission Techniques Panel Meeting, Memo No. 69-FM46-107, dated 17 April 1969.
- 7.4 Tindall, H. W., Lunar Rendezvous Mission Techniques; Memo No. 68-PA-T-219A, Dated 15 October 1968.
- 7.5 Guidance System Operations Plan for Manned LM Earth Orbital and Lunar Missions Using Program Luminay 1A, Section Four, PGNCS Operational Modes, Revision 1, dated April 1969.
- 7.6 GAP Assemble Revision 097 of AGS Program Luminay by NASA 2021112-041, dated 11 April 1969.
- 7.7 Grega, S. P.; LM Rendezvous Procedures D Mission, Final Revision A; dated 31 January 1969.
- 7.8 Hutchins, J. E.; LM Rendezvous Procedures F Mission, Final Revision A; dated 28 April 1969.
- 7.9 Apollo 11 Flight Plan AS-506/CSM-107/LM-5 Flight Planning Branch, Flight Crew Support Division, dated 15 April 1969.
- 7.10 Apollo Mission Techniques Mission F and G Rendezvous, Volume 1, dated 28 February 1969.
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- 7.12 Apollo 11 Flight Mission Rules, SA-506/CSM-107/LM-5, Flight Control Division, dated 16 April 1969.
- 7.13 LM/AGS Operating Manual Flight Program 6, TRW Document 11176-6033-T000, dated 6 April 1969.
- 7.14 CSM/LM Spacecraft Operational Data Book Volumes II, III, & IV, LM Data Book, SNA-8-D-027 with Revisions through 2 April 1969.